

# **Multi-Point Constraints**



### Multi-Point Constraints

- Single point constraint examples
  - $u_{x4} = 0$  linear, homogeneous  $u_{y9} = 0.6$  linear, non-homogeneous

### >Multi-Point constraint examples

 $u_{x2} = \frac{1}{2}u_{y2}$   $u_{x2} - 2u_{x4} + u_{x6} = 0.25$  $(x_5 + u_{x5} - x_3 - u_{x3})^2 + (y_5 + u_{y5} - y_3 - u_{y3})^2 = 0$ 

nonlinear, homogeneous

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### Modelling of joints

Perfect connection ensured here





### Modelling of joints

Mismatch between DOFs of beams and 2D solid – beam is free to rotate (rotation not transmitted to 2D solid)





Perfect connection by artificially extending beam into 2D solid (Additional mass)



- Modelling of joints
- Using MPC equations





#### Creation of MPC equations for offsets





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Creation of MPC equations for offsets  $d_{6}$  $d_8$ α  $d_9$  $d_2$ 'd₁  $d_5$  $d_6 = d_1 + \alpha \, d_5$  or  $d_1 + \alpha \, d_5 - d_6 = 0$  $d_7 = d_2 - \alpha d_4$  or  $d_2 - \alpha d_4 - d_7 = 0$ 

 $d_8 = d_3$  or  $d_3 - d_8 = 0$  $d_9 = d_5$  or  $d_5 - d_9 = 0$ 

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#### Modelling of joints **i**

Similar for plate connected to 3D solid





### Modelling of symmetric boundary conditions





Enforcement of mesh compatibility

Use lower order shape function to interpolate

$$d_x = 0.5(1-\eta) d_1 + 0.5(1+\eta) d_3$$

$$d_y = 0.5(1-\eta) d_4 + 0.5(1+\eta) d_6$$

Substitute value of  $\eta$  at node 2

$$0.5 d_1 - d_2 + 0.5 d_3 = 0$$
$$0.5 d_4 - d_5 + 0.5 d_6 = 0$$



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Enforcement of mesh compatibility

Use shape function of longer element to interpolate

 $d_x = -0.5 \eta (1-\eta) d_1 + (1+\eta)(1-\eta) d_3 + 0.5 \eta (1+\eta) d_5$ 

Substituting the values of  $\eta$  for the two additional nodes

$$d_{2} = 0.25 \times 1.5 \ d_{1} + 1.5 \times 0.5 \ d_{3} - 0.25 \times 0.5 \ d_{5}$$

$$d_{4} = -0.25 \times 0.5 \ d_{1} + 0.5 \times 1.5 \ d_{3} + 0.25 \times 1.5 \ d_{5}$$
Quad
$$d_{9} \uparrow Quad$$

$$d_{9} \uparrow Quad$$

$$d_{10} \downarrow d_{5}$$

$$d_{10} \downarrow d_{10} \downarrow d_{10}$$

$$d_{10} \downarrow d_{10} \downarrow d_{10} \downarrow d_{10}$$

$$d_{10} \downarrow d_{10} \downarrow d_{10} \downarrow d_{10}$$

$$d_{10} \downarrow d_{10} \downarrow d_{10$$

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Enforcement of mesh compatibility

In *x* direction,

$$0.375 d_1 - d_2 + 0.75 d_3 - 0.125 d_5 = 0$$
$$-0.125 d_1 + 0.75 d_3 - d_4 + 0.375 d_5 = 0$$

In y direction,

0.375 
$$d_6$$
-  $d_7$ +0.75  $d_8$ - 0.125  $d_{10} = 0$   
-0.125  $d_6$ +0.75  $d_8$ -  $d_9$ + 0.375  $d_{10} = 0$ 





Modelling of constraints by rigid body attachment

$$d_{1} = q_{1}$$

$$d_{2} = q_{1} + q_{2} I_{1}$$

$$d_{3} = q_{1} + q_{2} I_{2}$$

$$d_{4} = q_{1} + q_{2} I_{3}$$

Eliminate  $q_1$  and  $q_2$ 

$$(I_2 / I_1 - 1) d_1 - (I_2 / I_1) d_2 + d_3 = 0$$
$$(I_3 / I_1 - 1) d_1 - (I_3 / I_1) d_2 + d_4 = 0$$



(DOF in *x* direction not considered)



- Sources of Multi-Point Constraints
  - > Skew displacement BCs
  - Coupling nonmatched FEM meshes
  - > Global-local and multiscale analysis
  - > Incompressibility



- MPC Application Methods
  - Master-Slave Elimination
  - > Penalty Function Augmentation
  - > Lagrange Multiplier Adjunction



Example 1D Structure to Illustrate MPCs



Multi-Point constraint:

$$u_2 = u_6$$
 or  $u_2 - u_6 = 0$ 

Linear homogeneous MPC



### Example *1D* Structure to Illustrate MPCs



Unconstrained master stiffness equations

$$\begin{bmatrix} K_{11} & K_{12} & 0 & 0 & 0 & 0 & 0 \\ K_{12} & K_{22} & K_{23} & 0 & 0 & 0 & 0 \\ 0 & K_{23} & K_{33} & K_{34} & 0 & 0 & 0 \\ 0 & 0 & K_{34} & K_{44} & K_{45} & 0 & 0 \\ 0 & 0 & 0 & K_{45} & K_{55} & K_{56} & 0 \\ 0 & 0 & 0 & 0 & K_{56} & K_{66} & K_{67} \\ 0 & 0 & 0 & 0 & 0 & K_{67} & K_{77} \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \\ u_4 \\ u_5 \\ u_6 \\ u_7 \end{bmatrix} = \begin{bmatrix} f_1 \\ f_2 \\ f_3 \\ f_4 \\ f_5 \\ f_6 \\ f_7 \end{bmatrix}$$

 $\mathbf{K}\mathbf{u} = \mathbf{f}$ 

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Master Slave Method for Example Structure

Recall:  $u_2 = u_6$  or  $u_2 - u_6 = 0$ 

or

Taking u as master:

$$\begin{bmatrix} u_{1} \\ u_{2} \\ u_{3} \\ u_{4} \\ u_{5} \\ u_{6} \\ u_{7} \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \\ u_{3} \\ u_{4} \\ u_{5} \\ u_{7} \end{bmatrix}$$



Forming the Modified Stiffness Equations

Unconstrained master stiffness equations:

Master-slave transformation:  $\mathbf{u} = \mathbf{T}\hat{\mathbf{u}}$ 

Congruential transformation:

 $\hat{\mathbf{K}} = \mathbf{T}^T \mathbf{K} \mathbf{T}$  $\hat{\mathbf{f}} = \mathbf{T}^T \mathbf{f}$ 

 $\mathbf{K}\mathbf{u} = \mathbf{f}$ 

Modified stiffness equations:

$$\hat{\mathbf{K}}\hat{\mathbf{u}}=\hat{\mathbf{f}}$$

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### Modified Stiffness Equations for Example Structure

 $u_2$  as master and  $u_6$  as slave DOF.

$-K_{11}$	$K_{12}$	0	0	0	0	$[u_1]$	$\int f_1$
<i>K</i> <sub>12</sub>	$K_{22} + K_{66}$	<i>K</i> <sub>23</sub>	0	<i>K</i> <sub>56</sub>	$K_{67}$	$u_2$	$f_2 + f_6$
0	$K_{23}$	<i>K</i> <sub>33</sub>	<i>K</i> <sub>34</sub>	0	0	$u_3$	$f_3$
0	0	<i>K</i> <sub>34</sub>	$K_{44}$	$K_{45}$	0	$u_4$	 $f_4$
0	$K_{56}$	0	$K_{45}$	$K_{55}$	0	$u_5$	$f_5$
0	$K_{67}$	0	0	0	$K_{77}$	$\lfloor u_7 \rfloor$	$f_7$



Modified Stiffness Equations for Example Structure

 $u_6$  as master and  $u_2$  as slave DOF.

$$\begin{bmatrix} K_{11} & 0 & 0 & 0 & K_{12} & 0 \\ 0 & K_{33} & K_{34} & 0 & K_{23} & 0 \\ 0 & K_{34} & K_{44} & K_{45} & 0 & 0 \\ 0 & 0 & K_{45} & K_{55} & K_{56} & 0 \\ K_{12} & K_{23} & 0 & K_{56} & K_{22} + K_{66} & K_{67} \\ 0 & 0 & 0 & 0 & K_{67} & K_{77} \end{bmatrix} \begin{bmatrix} u_1 \\ u_3 \\ u_4 \\ u_5 \\ u_6 \\ u_7 \end{bmatrix} = \begin{bmatrix} f_1 \\ f_3 \\ f_4 \\ f_5 \\ f_2 + f_6 \\ f_7 \end{bmatrix}$$

Although they are algebraically equivalent, the latter would be processed faster if a skyline solver is used for the modified equations.



Multiple MPCs

Suppose

 $u_2 - u_6 = 0,$   $u_1 + 4u_4 = 0,$   $2u_3 + u_4 + u_5 = 0$ 

take 3, 4 and 6 as slaves:

$$u_6 = u_2,$$
  $u_4 = -\frac{1}{4}u_1,$   $u_3 = -\frac{1}{2}(u_4 + u_5) = \frac{1}{8}u_1 - \frac{1}{2}u_5$ 

and put in matrix form:

$$\begin{bmatrix} u_{1} \\ u_{2} \\ u_{3} \\ u_{4} \\ u_{5} \\ u_{6} \\ u_{7} \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ \frac{1}{8} & 0 & -\frac{1}{2} & 0 \\ -\frac{1}{4} & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \\ u_{5} \\ u_{7} \end{bmatrix}$$

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Nonhomogeneous MPCs

$$u_2 - u_6 = 0.2$$

In matrix form



# $\mathbf{u} = \mathbf{T}\hat{\mathbf{u}} + \mathbf{g}$



Nonhomogeneous MPCs
modified system:  $\hat{\mathbf{K}} \hat{\mathbf{u}} = \hat{\mathbf{f}}$ 

in which: 
$$\hat{\mathbf{K}} = \mathbf{T}^T \mathbf{K} \mathbf{T}$$
,  $\hat{\mathbf{f}} = \mathbf{T}^T \mathbf{f} - \mathbf{T}^T \mathbf{K} \mathbf{g}$ 

#### For the example structure

$$\begin{bmatrix} K_{11} & K_{12} & 0 & 0 & 0 & 0 \\ K_{12} & K_{22} + K_{66} & K_{23} & 0 & K_{56} & K_{67} \\ 0 & K_{23} & K_{33} & K_{34} & 0 & 0 \\ 0 & 0 & K_{34} & K_{44} & K_{45} & 0 \\ 0 & K_{56} & 0 & K_{45} & K_{55} & 0 \\ 0 & K_{67} & 0 & 0 & 0 & K_{77} \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \\ u_4 \\ u_5 \\ u_7 \end{bmatrix} = \begin{bmatrix} f_1 \\ f_2 + f_6 - 0.2K_{66} \\ f_3 \\ f_4 \\ f_5 - 0.2K_{56} \\ f_7 - 0.2K_{67} \end{bmatrix}$$



### The General Case of MPCs

For implementation in general-purpose programs the master-slave method can be described as follows. The degrees of freedoms in  $\mathbf{u}$  are classified into three types: independent or uncommitted, masters and slaves.

$$\begin{bmatrix} \mathbf{K}_{uu} & \mathbf{K}_{um} & \mathbf{K}_{us} \\ \mathbf{K}_{um}^{T} & \mathbf{K}_{mm} & \mathbf{K}_{ms} \\ \mathbf{K}_{us}^{T} & \mathbf{K}_{ms}^{T} & \mathbf{K}_{ss} \end{bmatrix} \begin{bmatrix} \mathbf{u}_{u} \\ \mathbf{u}_{m} \\ \mathbf{u}_{s} \end{bmatrix} = \begin{bmatrix} \mathbf{f}_{u} \\ \mathbf{f}_{m} \\ \mathbf{f}_{s} \end{bmatrix}$$

The MPCs may be written in matrix form as

$$\mathbf{A}_m \mathbf{u}_m + \mathbf{A}_s \mathbf{u}_s = \mathbf{g} \quad \Longrightarrow \quad \mathbf{u}_s = -\mathbf{A}_s^{-1} \mathbf{A}_m \mathbf{u}_m + \mathbf{A}_s^{-1} \mathbf{g} = \mathbf{T} \mathbf{u}_m + \mathbf{g}$$

Inserting into the partitioned stiffness matrix and symmetrizing

$$\begin{bmatrix} \mathbf{K}_{uu} & \mathbf{K}_{um}\mathbf{T} \\ \mathbf{T}^{T}\mathbf{K}_{um}^{T} & \mathbf{T}^{T}\mathbf{K}_{mm}\mathbf{T} \end{bmatrix} \begin{bmatrix} \mathbf{u}_{u} \\ \mathbf{u}_{m} \end{bmatrix} = \begin{bmatrix} \mathbf{f}_{u} - \mathbf{K}_{us}\mathbf{g} \\ \mathbf{f}_{m} - \mathbf{K}_{ms}\mathbf{g} \end{bmatrix}$$

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#### 

- > exact if precautions taken
- ▶ easy to understand
- retains positive definiteness
- >important applications to model reduction

### DISADVANTAGES

- requires user decisions
- >messy implementation for general MPCs
- > sensitive to constraint dependence
- restricted to linear constraints

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### Penalty Function Method, Physical Interpretation

Recall the example structure



under the homogeneous MPC

$$u_2 = u_6$$

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### Penalty Function Method, Physical Interpretation





### Penalty Function Method, Physical Interpretation

Upon merging the penalty element the modified stiffness equations are

$-K_{11}$	$K_{12}$	0	0	0	0	0 7	$\int u_1$		$\lceil f_1 \rceil$
<i>K</i> <sub>12</sub>	$K_{22} + w$	<i>K</i> <sub>23</sub>	0	0	-w	0	$u_2$		$f_2$
0	<i>K</i> <sub>23</sub>	<i>K</i> <sub>33</sub>	<i>K</i> <sub>34</sub>	0	0	0	<i>u</i> <sub>3</sub>		$f_3$
0	0	<i>K</i> <sub>34</sub>	$K_{44}$	$K_{45}$	0	0	$u_4$	=	$f_4$
0	0	0	$K_{45}$	$K_{55}$	$K_{56}$	0	$u_5$		$f_5$
0	-w	0	0	$K_{56}$	$K_{66} + w$	<i>K</i> <sub>67</sub>	$u_6$		$f_6$
0	0	0	0	0	$K_{67}$	$K_{77}$	$\lfloor u_7 \rfloor$		$\lfloor f_7 \rfloor$

This modified system is submitted to the equation solver. Note that **u** retains the same arrangement of DOFs. Penalty Function Method - General MPCs

Premultiply both sides by  $\begin{bmatrix} 3 & 1 & -4 \end{bmatrix}^T$ 

$$\begin{bmatrix} 9 & 3 & -12 \\ 3 & 1 & -4 \\ -12 & -4 & 16 \end{bmatrix} \begin{bmatrix} u_3 \\ u_5 \\ u_6 \end{bmatrix} = \begin{bmatrix} 3 \\ 1 \\ -4 \end{bmatrix}$$

Scale by *w* and merge:

$$\begin{bmatrix} K_{11} & K_{12} & 0 & 0 & 0 & 0 & 0 \\ K_{12} & K_{22} & K_{23} & 0 & 0 & 0 & 0 \\ 0 & K_{23} & K_{33} + 9w & K_{34} & 3w & -12w & 0 \\ 0 & 0 & K_{34} & K_{44} & K_{45} & 0 & 0 \\ 0 & 0 & 3w & K_{45} & K_{55} + w & K_{56} - 4w & 0 \\ 0 & 0 & -12w & 0 & K_{56} - 4w & K_{66} + 16w & K_{67} \\ 0 & 0 & 0 & 0 & 0 & K_{67} & K_{77} \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \\ u_4 \\ u_5 \\ u_6 \\ u_7 \end{bmatrix} = \begin{bmatrix} f_1 \\ f_2 \\ f_3 + 3w \\ f_4 \\ f_5 + w \\ f_6 - 4w \\ f_7 \end{bmatrix}$$

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Theory of Penalty Function Method - General MPCs

$$\mathbf{t} = \mathbf{C}\mathbf{U} - \mathbf{Q} \qquad \text{(Constrain equations)}$$
$$\boldsymbol{\Pi}_{p} = \frac{1}{2}\mathbf{U}^{T}\mathbf{K}\mathbf{U} - \mathbf{U}^{T}\mathbf{F} + \frac{1}{2}\mathbf{t}^{T}\boldsymbol{\alpha}\mathbf{t}$$

 $\alpha = [\alpha_1 \ \alpha_2 \ \dots \ \alpha_m]$  is a diagonal matrix of 'penalty numbers'

stationary condition of the modified functional requires the derivatives of  $\Pi_p$  with respect to the  $\mathbf{U}_i$  to vanish

$$\frac{\partial \Pi_p}{d\mathbf{U}} = 0 \quad \rightarrow \quad \mathbf{K}\mathbf{U} - \mathbf{F} + \mathbf{C}^T \boldsymbol{\alpha}\mathbf{C}\mathbf{U} + \mathbf{C}^T \boldsymbol{\alpha}\mathbf{Q} = \mathbf{0}$$

$$[\mathbf{K} + \mathbf{C}^T \boldsymbol{\alpha} \mathbf{C}]\mathbf{U} = \mathbf{F} + \mathbf{C}^T \boldsymbol{\alpha} \mathbf{Q}$$

Penalty matrix

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Theory of Penalty Function Method - General MPCs



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Assessment of Penalty Function Method

### ADVANTAGES

- > general application (inc' nonlinear MPCs)
- > easy to implement using FE library and standard assembler
- > no change in vector of unknowns
- retains positive definiteness
- insensitive to constraint dependence

### DISADVANTAGES

- > selection of weight left to user
- > accuracy limited by ill-conditioning
- > the constraint equations can only be satisfied approximately.



Lagrange Multiplier Method, Physical Interpretation





### Lagrange Multiplier Method

Because  $\lambda$  is unknown, it is passed to the LHS and appended to the node-displacement vector:



This is now a system of 7 equations and 8 unknowns. Needs an extra equation: the MPC.

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### Lagrange Multiplier Method

Append MPC as additional equation:

$\begin{bmatrix} I \end{bmatrix}$	$K_{11}$	<i>K</i> <sub>12</sub>	0	0	0	0	0	0	$\begin{bmatrix} u_1 \end{bmatrix}$		$\lceil f_1 \rceil$	
	$K_{12}$	<i>K</i> <sub>22</sub>	<i>K</i> <sub>23</sub>	0	0	0	0	1	$u_2$		$f_2$	
	0	<i>K</i> <sub>23</sub>	<i>K</i> <sub>33</sub>	<i>K</i> <sub>34</sub>	0	0	0	0	$u_3$		$f_3$	
	0	0	<i>K</i> <sub>34</sub>	$K_{44}$	$K_{45}$	0	0	0	$u_4$		$f_4$	
	0	0	0	$K_{45}$	$K_{55}$	$K_{56}$	0	0	$u_5$	_	$f_5$	
	0	0	0	0	$K_{56}$	$K_{66}$	$K_{67}$	-1	$u_6$		$f_6$	
	0	0	0	0	0	$K_{67}$	$K_{77}$	0	$u_7$		$f_7$	
L	0	1	0	0	0	-1	0	0	λ		0	

This is the *multiplier-augmented system*. The new coefficient matrix is called the *bordered stiffness*.



### IMPLEMENTATION OF MPC EQUATIONS

 $\mathbf{KU} = \mathbf{F}$  (Global system equation)

(Matrix form of MPC equations)

Constant matrices

 $\mathbf{CU} - \mathbf{Q} = \mathbf{0}$ 

Optimization problem for solution of nodal degrees of freedom:

Find U to Minimize:  $\Pi_{p} = \frac{1}{2} \mathbf{U}^{T} \mathbf{K} \mathbf{U} - \mathbf{U}^{T} \mathbf{F}$ Subject to:  $\mathbf{C}\mathbf{U} - \mathbf{Q} = \mathbf{0}$ 



Lagrange multiplier method

- $\boldsymbol{\lambda} = \begin{bmatrix} \lambda_1 & \lambda_2 & \cdots & \lambda_m \end{bmatrix}^T$  (Lagrange multipliers)
- $\boldsymbol{\lambda}^{T} \{ \mathbf{C}\mathbf{U} \mathbf{Q} \} = 0 \qquad \text{Multiplied to MPC equations}$
- Find U and  $\lambda$  to Minimize:  $L = \frac{1}{2} \mathbf{U}^T \mathbf{K} \mathbf{U} - \mathbf{U}^T \mathbf{F} + \lambda^T \{\mathbf{C} \mathbf{U} - \mathbf{Q}\}$  Added to functional The stationary condition requires the derivatives of *L* with respect to the U<sub>i</sub> and  $\lambda_i$  to vanish.

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### Lagrange Multiplier Method - Multiple MPCs

Three MPCs: <i>u</i>	$u_2 - u_6 =$	0,	$5u_2$	2 - 8i	$u_7 =$	3,	3и	<i>u</i> <sub>3</sub> +	$u_5 -$	$4u_6$	= 1	
Recipe step #1: append the 3 constraints	$\begin{bmatrix} K_{11} \\ K_{12} \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix}$	$ \begin{array}{c} K_{12} \\ K_{22} \\ K_{23} \\ 0 \\ 0 \\ 0 \\ 0 \\ 1 \\ 5 \\ 0 \\ \end{array} $	$ \begin{array}{c} 0\\ K_{23}\\ K_{33}\\ K_{34}\\ 0\\ 0\\ 0\\ 0\\ 0\\ 3 \end{array} $	$egin{array}{c} 0 \\ 0 \\ K_{34} \\ K_{44} \\ K_{45} \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{array}$	$ \begin{array}{c} 0 \\ 0 \\ K_{45} \\ K_{55} \\ K_{56} \\ 0 \\ 0 \\ 1 \end{array} $	$ \begin{array}{c} 0 \\ 0 \\ 0 \\ K_{56} \\ K_{66} \\ K_{67} \\ -1 \\ 0 \\ -4 \end{array} $	$ \begin{array}{c} 0 & -8 \\ 0 \\ 0 \\ 0 \\ K_{67} \\ K_{77} \end{array} $		$\begin{bmatrix} u_1 \\ u_2 \\ u_3 \\ u_4 \\ u_5 \\ u_6 \\ u_7 \end{bmatrix} =$	$\begin{bmatrix} f_1 \\ f_2 \\ f_3 \\ f_4 \\ f_5 \\ f_6 \\ f_7 \\ 0 \\ 3 \\ 1 \end{bmatrix}$		
Recipe step #2: append multipliers symmetrize and fill	$\mathbf{S}, \qquad \begin{bmatrix} K_{11} \\ K_{12} \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix}$	$ \begin{array}{c} K_{12} \\ K_{22} \\ K_{23} \\ 0 \\ 0 \\ 0 \\ 0 \\ 1 \\ 5 \\ 0 \\ \end{array} $	$ \begin{array}{c} 0\\ K_{23}\\ K_{33}\\ K_{34}\\ 0\\ 0\\ 0\\ 0\\ 0\\ 0\\ 3 \end{array} $	$ \begin{array}{c} 0\\ 0\\ K_{34}\\ K_{44}\\ K_{45}\\ 0\\ 0\\ 0\\ 0\\ 0\\ 0\\ 0\\ 0\\ 0\\ 0\\ 0\\ 0\\ 0\\$	$\begin{array}{c} 0 \\ 0 \\ 0 \\ K_{45} \\ K_{55} \\ K_{56} \\ 0 \\ 0 \\ 0 \\ 1 \end{array}$	$ \begin{array}{c} 0 \\ 0 \\ 0 \\ K_{56} \\ K_{66} \\ K_{67} \\ -1 \\ 0 \\ -4 \end{array} $	$ \begin{array}{c} 0 \\ 0 \\ 0 \\ 0 \\ K_{67} \\ K_{77} \\ 0 \\ -8 \\ 0 \\ 0 \end{array} $	0 1 0 0 -1 0 0 0 0 0	$ \begin{array}{c} 0 \\ 5 \\ 0 \\ 0 \\ 0 \\ -8 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{array} $	$ \begin{array}{c} 0 & - \\ 0 \\ 3 \\ 0 \\ 1 \\ -4 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ -4 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0$	$\begin{bmatrix} u_1 \\ u_2 \\ u_3 \\ u_4 \\ u_5 \\ u_6 \\ u_7 \\ \lambda_1 \\ \lambda_2 \\ \lambda_3 \end{bmatrix}$	$ \begin{array}{c} f_{1} \\ f_{2} \\ f_{3} \\ f_{4} \\ f_{5} \\ f_{6} \\ f_{7} \\ 0 \\ 3 \\ 1 \end{array} $

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Example: Five bar truss with inclined support



E = 70 Gpa,  $A = 10^{-3}$  m<sup>2</sup>, P = 20 kN.

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### Example: Five bar truss with inclined support

Equations for element 1



### Example: Five bar truss with inclined support

Equations for element 2

E = 70000 A = 1000Element node Global node number Х 5 0 0 4 5000. 3000. 2  $x_1 = 0$   $y_1 = 0$   $x_2 = 5000.$   $y_2 = 3000.$  $L = \sqrt{(x_2 - x_1)^2 + (y_2 - y_1)^2} = 5830.95$ 0 Direction cosines:  $\ell_s = \frac{x_2 - x_1}{L} = 0.857493$   $m_s = \frac{y_2 - y_1}{L} = 0.514496$  $\mathbf{k}^{(2)} = \begin{pmatrix} 8827.13 & 5296.28 & -8827.13 & -5296.28 \\ 5296.28 & 3177.77 & -5296.28 & -3177.77 \\ -8827.13 & -5296.28 & 8827.13 & 5296.28 \\ -5296.28 & -3177.77 & 5296.28 & 3177.77 \end{pmatrix}$ 

(m)

### Example: Five bar truss with inclined support

Equations for element 3

E = 70000 A = 1000Element node Global node number x 2  $x_1 = 0$   $y_1 = 0$   $x_2 = 0$   $y_2 = 3000.$  $L = \sqrt{(x_2 - x_1)^2 + (y_2 - y_1)^2} = 3000.$ Direction cosines:  $\ell_s = \frac{x_2 - x_1}{L} = 0$   $m_s = \frac{y_2 - y_1}{L} = 1.$  $\mathbf{k}^{(3)} = \begin{pmatrix} 0. & 0. & 0. & 0. \\ 0. & 23333.3 & 0. & -23333.3 \\ 0. & 0. & 0. & 0. \\ 0. & -23333.3 & 0. & 23333.3 \end{pmatrix}$ 



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### Example: Five bar truss with inclined support

Equations for element 4

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### Example: Five bar truss with inclined support

Equations for element 4

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4

2

Δ

Δ

### Example: Five bar truss with inclined support

Equations for element 5

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### Example: Five bar truss with inclined support

( 1	7654.3	0	0	0	-8827.13	5296.28	-8827.13	-5296.28	$\left( \begin{array}{c} u_1 \end{array} \right)$	1	(0)
	0	29688.9	0	-23333.3	5296.28	-3177.77	-5296.28	-3177.77	$\mathbf{v}_1$		0
	0	0	14000.	0	0	0	-14000.	0	<b>u</b> <sub>2</sub>		0
	0	-23333.3	0	23333.3	0	0	0	0	$\mathbf{v}_2$		0
-	8827.13	5296.28	0	0	8827.13	-5296.28	0	0	u <sub>3</sub>	=	20000.
	5296.28	-3177.77	0	0	-5296.28	14844.4	0	-11666.7	<b>v</b> <sub>3</sub>		0
-	8827.13	-5296.28	-14000.	0	0	0	22827.1	5296.28	u <sub>4</sub>		0
( –	5296.28	-3177.77	0	0	0	-11666.7	5296.28	14844.4	$\left( v_{4} \right)$		$\left( \begin{array}{c} 0 \end{array} \right)$

Essential boundary conditions

Node	dof	Value
2	$u_2$	0
2	$\mathbf{v}_2$	0

### After adjusting for essential boundary conditions

(	17654.3	0	-8827.13	5296.28	-8827.13	-5296.28	$\left( u_{1}\right)$		$\begin{pmatrix} 0 \end{pmatrix}$
	0	29688.9	5296.28	-3177.77	-5296.28	-3177.77	$\mathbf{v}_1$		0
	-8827.13	5296.28	8827.13	-5296.28	0	0	u <sub>3</sub>		20000.
	5296.28	-3177.77	-5296.28	14844.4	0	-11666.7	<b>V</b> <sub>3</sub>	=	0
	-8827.13	-5296.28	0	0	22827.1	5296.28	u4		0
	-5296.28	-3177.77	0	-11666.7	5296.28	14844.4	$\left( v_{4} \right)$		$\left(\begin{array}{c}0\end{array}\right)$

Multipoint constraint due to inclined support at node 1:  $u_1 \sin(\pi/6) + v_1 \cos(\pi/6) = 0$ 

The augmented global equations with the Lagrange multiplier are as follows.



Solving the final system of global equations we get

$$\{u_1 = 5.14286, v_1 = -2.96923, u_3 = 16.8629, v_3 = 12.788, u_4 = -1.42857, v_4 = 11.7594, \lambda = 80000.\}$$

Solution for element 1

Nodal coordinates

Element node Global node number x y  
1 1 1 0 0  
2 3 5000. -3000.  

$$x_1 = 0$$
  $y_1 = 0$   $x_2 = 5000.$   $y_2 = -3000.$   
 $L = \sqrt{(x_2 - x_1)^2 + (y_2 - y_1)^2} = 5830.95$   
Direction cosines:  $\ell_s = \frac{x_2 - x_1}{L} = 0.857493$   $m_s = \frac{y_2 - y_1}{L} = -0.514496$   
Global to local transformation matrix  
( 0.857493 -0.514496 0 0)

$$T = \left(\begin{array}{ccc} 0 & 0 & 0 & 0.857493 & -0.514496 \end{array}\right)$$

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Element nodal displacements in global coordinates

$$d = \begin{pmatrix} u_1 \\ v_1 \\ u_3 \\ v_3 \end{pmatrix} = \begin{pmatrix} 5.14286 \\ -2.96923 \\ 16.8629 \\ 12.788 \end{pmatrix}$$
Element nodal displacements  
in local coordinates  
16.8629 (7.88048)

E = 70000 A = 1000

Axial strain,  $\epsilon = (d_2 - d_1)/L = 0.000333197$ Stress Axial force 1 23.3238 23323.8 Axial stress,  $\sigma = E\epsilon = 23.3238$ 2 23.3238 23323.8 Axial force =  $\sigma A = 23323.8$ 3 69.282 69282. 4 -20.-20000.5 -12.-12000.وانشكام صنمتي اصفهان- وانشكوم 50 روش اجزای محدود مكانيكر



### Assessment of Lagrange Multiplier Method

### ADVANTAGES

- General application
- Constraint equations are satisfied exactly

### DISADVANTAGES

- Difficult implementation
- >Total number of unknowns is increased
- > Expanded stiffness matrix is non-positive definite due to the presence of zero diagonal terms
- > Efficiency of solving the system equations is lower
- > sensitive to constraint dependence

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### MPC Application Methods: Assessment Summary

	Master-Slave Elimination	Penalty Function	Lagrange Multiplier		
Generality	fair	excellent	excellent		
Ease of implementation	poor to fair	good	fair		
Sensitivity to user decisions	high	high	small to none		
Accuracy	variable	mediocre	excellent		
Sensitivity as regards constraint dependence	high	none	high		
Retains positive definiteness	yes	yes	no		



References

1- Finite Element Method: A Practical Course by: S. S. Quek, G.R. Liu, 2003.

2- Introduction to Finite Element Methods, by: Carlos Felippa, University of Colorado at Boulder .
 http://www.colorado.edu/engineering/cas/courses.d/IFEM.d/