

III. STRESS-STRAIN RELATIONS. GENERALIZED HOOKE'S LAW

3.1 DIRECT FORM. INVERSE FORM.

$$\epsilon_{ij} = A_{ijkl} \sigma_{kl}, \quad i, j, k, l = 1, 2, 3 \quad (3.1-1)$$

where

$$A_{ijkl} = A_{jikl} = A_{ijlk} = A_{klij} \quad (3.1-2)$$

e.g.

$$\begin{aligned} \epsilon_{11} = & A_{1111} \sigma_{11} + A_{1112} \sigma_{12} + A_{1113} \sigma_{13} \\ & + A_{1121} \sigma_{21} + A_{1122} \sigma_{22} + A_{1123} \sigma_{23} \\ & + A_{1131} \sigma_{31} + A_{1132} \sigma_{32} + A_{1133} \sigma_{33} \end{aligned} \quad (3.1-3)$$

or alternatively,

$$\begin{matrix} \begin{pmatrix} \epsilon_{11} \\ \epsilon_{22} \\ \epsilon_{33} \\ \epsilon_{12} \\ \epsilon_{13} \\ \epsilon_{23} \end{pmatrix} \\ 6 \times 1 \end{matrix} = \begin{matrix} \begin{bmatrix} A_{1111} & A_{1122} & A_{1133} & A_{1112} & A_{1113} & A_{1123} \\ A_{2211} & A_{2222} & A_{2233} & A_{2212} & A_{2213} & A_{2223} \\ A_{3311} & A_{3322} & A_{3333} & A_{3312} & A_{3313} & A_{3323} \\ A_{1211} & A_{1222} & A_{1233} & A_{1212} & A_{1213} & A_{1223} \\ A_{1311} & A_{1322} & A_{1333} & A_{1312} & A_{1313} & A_{1323} \\ A_{2311} & A_{2322} & A_{2333} & A_{2312} & A_{2313} & A_{2323} \end{bmatrix} \\ 6 \times 6 \end{matrix} \begin{matrix} \begin{pmatrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{33} \\ 2\sigma_{12} \\ 2\sigma_{13} \\ 2\sigma_{23} \end{pmatrix} \\ 6 \times 1 \end{matrix} \quad (3.1-4)$$

Inverse Form

$$\sigma_{ij} = C_{ijkl} \epsilon_{kl} \quad i, j, k, l = 1, 2, 3 \quad (3.1-5)$$

where

$$C_{ijkl} = C_{jikl} = C_{ijlk} = C_{klij} \quad (3.1-6)$$

e.g.

$$\begin{aligned} \sigma_{12} = & C_{1211}\epsilon_{11} + C_{1212}\epsilon_{12} + C_{1213}\epsilon_{13} \\ & + C_{1221}\epsilon_{21} + C_{1222}\epsilon_{22} + C_{1223}\epsilon_{23} \\ & + C_{1232}\epsilon_{31} + C_{1232}\epsilon_{32} + C_{1233}\epsilon_{33} \end{aligned} \quad (3.1-7)$$

or alternatively,

$$\begin{matrix} \left[ \begin{matrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{33} \\ \sigma_{12} \\ \sigma_{13} \\ \sigma_{23} \end{matrix} \right]_{6 \times 1} = \left[ \begin{matrix} C_{1111} & C_{1122} & C_{1133} & C_{1112} & C_{1113} & C_{1123} \\ C_{2211} & C_{2222} & C_{2233} & C_{2212} & C_{2213} & C_{2223} \\ C_{3311} & C_{3322} & C_{3333} & C_{3312} & C_{3313} & C_{3323} \\ C_{1211} & C_{1222} & C_{1233} & C_{1212} & C_{1213} & C_{1223} \\ C_{1311} & C_{1322} & C_{1333} & C_{1312} & C_{1313} & C_{1323} \\ C_{2311} & C_{2322} & C_{2333} & C_{2312} & C_{2313} & C_{2323} \end{matrix} \right]_{6 \times 6} \left[ \begin{matrix} \epsilon_{11} \\ \epsilon_{22} \\ \epsilon_{33} \\ \epsilon_{12} \\ \epsilon_{13} \\ \epsilon_{23} \end{matrix} \right]_{6 \times 1} \end{matrix} \quad (3.1-8)$$

3.2 SPECIAL SYMMETRY

Upon coordinate transformation, Fig.A.1-2.

$$A'_{ijkl} = a_{ip} a_{jq} a_{kr} a_{ls} A_{pqrs} \quad (3.2-1)$$

where  $a_{ip}$ , etc., are direction cosines between the  $\bar{x}'$  and the  $\bar{x}$  coordinates.

Symmetry With Respect to One Plane: monoclinic material, Fig.3.2-1

$a_{ij}$	$x'_1$	$x'_2$	$x'_3$
$x_1$	1	0	0
$x_2$	0	1	0
$x_3$	0	0	-1

The stiffness matrix becomes:

$$\begin{bmatrix}
 A_{1111} & A_{1122} & A_{1133} & A_{1112} & 0 & 0 \\
 A_{1122} & A_{2222} & A_{2233} & A_{2212} & 0 & 0 \\
 A_{1133} & A_{2233} & A_{3333} & A_{3312} & 0 & 0 \\
 A_{1112} & A_{2212} & A_{3312} & A_{1212} & 0 & 0 \\
 0 & 0 & 0 & 0 & A_{1313} & A_{1323} \\
 0 & 0 & 0 & 0 & A_{1323} & A_{2323}
 \end{bmatrix} \quad (3.2-2)$$

The moduli matrix  $C_{ijkl}$  are similar.

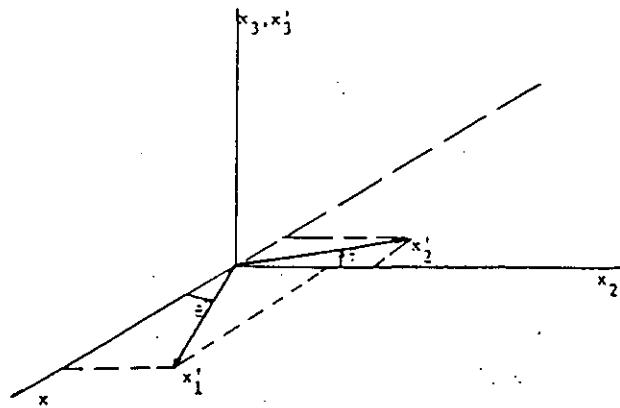


Fig.3.2-3 Transversely isotropic system

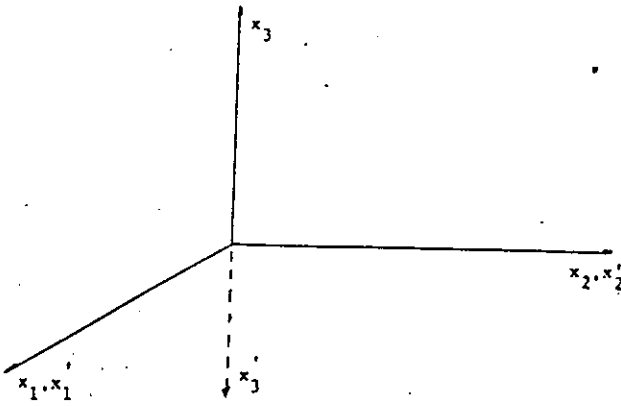


Fig.3.2-1 Monodinic system

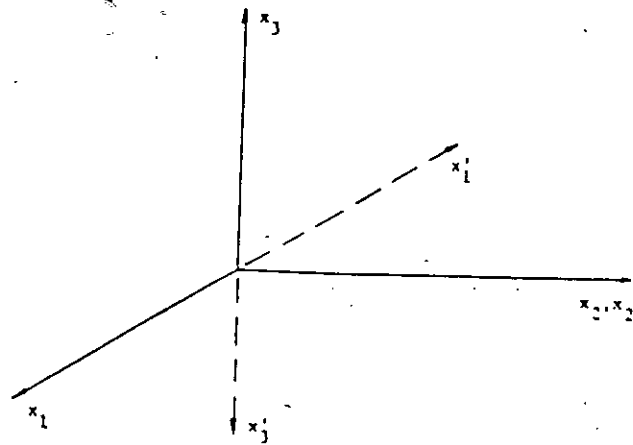


Fig.3.2-2 Orthotropic system

Symmetry With Respect to Two Orthogonal Planes: orthotropic material,

Fig. 3.2-2.

$a_{ij}$	$x'_1$	$x'_2$	$x'_3$
$x_1$	-1	0	0
$x_2$	0	1	0
$x_3$	0	0	-1

The stiffness matrix becomes:

$$\begin{bmatrix}
 A_{1111} & A_{1122} & A_{1133} & 0 & 0 & 0 \\
 A_{1122} & A_{2222} & A_{2233} & 0 & 0 & 0 \\
 A_{1133} & A_{2233} & A_{3333} & 0 & 0 & 0 \\
 0 & 0 & 0 & A_{1212} & 0 & 0 \\
 0 & 0 & 0 & 0 & A_{1313} & 0 \\
 0 & 0 & 0 & 0 & 0 & A_{2323}
 \end{bmatrix} \quad (3.2-3)$$

The moduli matrix  $C_{ijkl}$  is similar.

Symmetry With Respect to One Axis: transversely isotropic or cross anisotropic material, Fig. 3.2-3.

$a_{ij}$	$x'_1$	$x'_2$	$x'_3$
$x_1$	$\cos \theta$	$-\sin \theta$	0
$x_2$	$\sin \theta$	$\cos \theta$	0
$x_3$	0	0	1

The stiffness matrix is:

$$\begin{bmatrix}
 A_{1111} & A_{1122} & A_{1133} & 0 & 0 & 0 \\
 A_{1122} & A_{1111} & A_{1133} & 0 & 0 & 0 \\
 A_{1133} & A_{1133} & A_{3333} & 0 & 0 & 0 \\
 0 & 0 & 0 & \frac{1}{2}(A_{1111} - A_{1122}) & 0 & 0 \\
 0 & 0 & 0 & 0 & A_{1313} & 0 \\
 0 & 0 & 0 & 0 & 0 & A_{1313}
 \end{bmatrix} \quad (3.2-4)$$

The compliance matrix is similar in form.

Rotational Symmetry With Respect to Two Mutually Perpendicular Axes:

isotropic material

The stiffness matrix is:

$$\begin{bmatrix}
 A_{1111} & A_{1122} & A_{1122} & 0 & 0 & 0 \\
 A_{1122} & A_{1111} & A_{1122} & 0 & 0 & 0 \\
 A_{1122} & A_{1122} & A_{1111} & 0 & 0 & 0 \\
 0 & 0 & 0 & \frac{1}{2}(A_{1111} - A_{1122}) & 0 & 0 \\
 0 & 0 & 0 & 0 & \frac{1}{2}(A_{1111} - A_{1122}) & 0 \\
 C & 0 & 0 & 0 & 0 & \frac{1}{2}(A_{1111} - A_{1122})
 \end{bmatrix} \quad (3.2-5)$$

The compliance matrix takes a form that is similar to the above.

### 3.3 CONVENTIONAL ELASTIC CONSTANTS FOR ISOTROPIC MATERIAL.

$\lambda, \mu$	Lame' constants
E	Young's modulus
G	shear modulus, $G = \mu$
K	bulk modulus
$\nu$	Poisson's ratio

$$A_{1122} = \lambda, A_{1212} = \frac{1}{2}(A_{1111} - A_{1122}), A_{1111} = \lambda + 2\mu \quad (3.3-1)$$

$$\begin{aligned} C_{1111} &= (\lambda + \mu)/\mu (3\lambda + 2\mu) \\ C_{1122} &= -\lambda/2\mu (3\lambda + 2\mu) \\ \frac{1}{2}(C_{1111} - C_{1122}) &= 1/4\mu \end{aligned} \quad (3.3-2)$$

### 3.4 STRESS-STRAIN RELATIONS. ISOTROPIC MATERIALS

$$\epsilon_{ij} = \sigma_{ij}/2\mu - [\lambda/2\mu(3\lambda+2\mu)] \sigma_{kk} \delta_{ij} \quad (3.4-1)$$

$$\sigma_{ij} = 2\mu \epsilon_{ij} + \lambda \epsilon_{kk} \delta_{ij} \quad (3.4-2)$$

#### Rectangular Co-ordinates (x, y, z)

Direct form:

$$\sigma = \frac{[\sigma_{11} + \sigma_{22} + \sigma_{33}]}{3}$$

$$2\mu \epsilon_{xx} = \sigma_{xx} - 3[\nu/(1+\nu)] \sigma \quad (3.4-3a)$$

$$2\mu \epsilon_{yy} = \sigma_{yy} - 3[\nu/(1+\nu)] \sigma \quad (3.4-3b)$$

$$2\mu \epsilon_{zz} = \sigma_{zz} - 3[\nu/(1+\nu)] \sigma \quad (3.4-3c)$$

$$2\mu \epsilon_{xy} = \sigma_{xy} \quad (3.4-3d)$$

$$2\mu \epsilon_{yz} = \sigma_{yz} \quad (3.4-3e)$$

$$2\mu \epsilon_{zx} = \sigma_{zx} \quad (3.4-3f)$$

where

$$\sigma = [\sigma_{xx} + \sigma_{yy} + \sigma_{zz}]/3 \quad (3.4-4)$$

### 3.5 PROBLEMS

1. An elastic layer, of moduli  $E$  and  $\nu$ , is enclosed in a rigid container with one end open. The layer is then compressed by a rigid stamp which applies a direct stress  $\sigma_z$  to the layer. Let the apparent Young's modulus be  $E'$  (that is,  $\sigma_z/\epsilon_z$ ). Plot the ratio  $E'/E$  as a function of  $\nu$ , knowing that  $0 \leq \nu \leq 0.5$ .

2. Prove that Eq.(3.5-1) follows from Eqs.(3.5-2),(3.5-3)

$$e = \epsilon_{xx} + \epsilon_{yy} + \epsilon_{zz} = [(1-2\nu)/E](\sigma_{xx} + \sigma_{yy} + \sigma_{zz}) \quad (3.5-1)$$

$$\left. \begin{aligned} \sigma_{xx} &= \lambda e + 2\mu\epsilon_{xx} \\ \sigma_{yy} &= \lambda e + 2\mu\epsilon_{yy} \\ \sigma_{zz} &= \lambda e + 2\mu\epsilon_{zz} \end{aligned} \right\} \quad (3.5-2)$$

$$\mu = E/[2(1+\nu)], \quad \lambda = \nu E/[(1+\nu)(1-2\nu)] \quad (3.5-3)$$

3. Show that the assumption that an elastic isotropic solid is incompressible is equivalent to the assumption that Poisson's ratio is equal to one half.

\*4. Prove that the principal axes of stress coincide with the principal axes of strain for a linear, isotropic, elastic solid, and relate the principal values to each other.



5. Show that the multi-axial isotropic Hooke's law can be obtained by the following argument: An elementary rectangular parallelepiped subjected to tensile stresses  $\sigma_{xx}$  on opposite faces will experience a longitudinal extension  $\epsilon_{xx} = \sigma_{xx}/E$  and lateral contractions  $\epsilon_{yy} = \epsilon_{zz} = -\nu\epsilon_{xx}$ . Now consider the effect of stresses  $\sigma_{xx}$ ,  $\sigma_{yy}$ ,  $\sigma_{zz}$  and superpose the resulting strains.

6. Show that the index notation of the isotropic Hooke's law may be written as

$$\sigma_{ij} = 2\mu[\epsilon_{ij} + \frac{\nu}{1-2\nu} \epsilon_{kk} \delta_{ij}] \quad (3.5-4)$$

7. Show directly from the generalized Hooke's law that in an isotropic body the principal axes of strain coincide with those of stress. Hint: Take the coordinate axes along the principal axes of strain and consider the effect on  $\sigma_{23}$  and  $\sigma_{31}$  of a rotation of axes by  $180^\circ$  about the  $x_3$ -axis.

8. For an isotropic linear elastic solid, derive the following relations between the Lamé constants  $\lambda$  and  $\mu$ , Poisson's ratio  $\nu$ , Young's modulus  $E$ , and the bulk modulus  $K$ :

$$\lambda = \frac{2\mu\nu}{1-2\nu} = \frac{\nu(E-2\mu)}{3\mu-E} = K - \frac{2}{3}\mu = \frac{E\nu}{(1+\nu)(1-2\nu)}$$

$$= \frac{3K\nu}{1+\nu} = \frac{3K(3K-E)}{9K-E}$$

$$\mu = \frac{\lambda(1-2\nu)}{2\nu} - \frac{3}{2}(K-\lambda) - \frac{E}{2(1+\nu)} - \frac{3K(1-2\nu)}{2(1+\nu)}$$

$$= \frac{3KE}{9K-E}$$

$$\nu = \frac{\lambda}{2(\lambda+\mu)} - \frac{\lambda}{3K-\lambda} - \frac{E}{2\mu} - 1 - \frac{3K-2\mu}{2(3K+\mu)}$$

$$= \frac{3K-E}{6K}$$

$$E = \frac{\mu(3\lambda+2\mu)}{\lambda+\mu} - \frac{\lambda(1+\nu)(1-2\nu)}{\nu} - \frac{9K(K-\lambda)}{3K-\lambda}$$

$$= 2\mu(1+\nu) - \frac{9K\mu}{3K+\mu} - 3K(1-2\nu)$$

$$K = \lambda + \frac{2}{3}\mu = (\lambda+E)/2 + \sqrt{(E-3\lambda)^2 + 8\lambda E}/6$$

$$= \lambda(1+\nu)/(3\nu) - \mu E/[3(3\mu-E)]$$

$$= 2\mu(1+\nu)/[3(1-2\nu)] - E/[3(1-2\nu)]$$

TABLE 3.3-1 INTERRELATION BETWEEN THE ELASTIC CONSTANTS OF ISOTROPIC HOMOGENEOUS BODY

	$\lambda$	$\mu$	E	$\nu$	K
$\lambda, \mu$			$\frac{\mu(3\lambda+2\mu)}{\lambda+\mu}$	$\frac{\lambda}{2(\lambda+\mu)}$	$\frac{3\lambda+2\mu}{3}$
$\lambda, E$		$\frac{(E-3\lambda)+A^*}{4}$		$\frac{-(E+\lambda)+A^*}{4\lambda}$	$\frac{3(\lambda+E)+A^*}{6}$
$\lambda, \nu$		$\frac{\lambda(1-2\nu)}{2\nu}$	$\frac{\lambda(1+\nu)(1-2\nu)}{\nu}$		$\frac{\lambda(1+\nu)}{3\nu}$
$\lambda, K$		$\frac{3(K-\lambda)}{2}$	$\frac{9K(K-\lambda)}{3K-\lambda}$	$\frac{\lambda}{3K-\lambda}$	
$\mu, E$	$\frac{\mu(2\mu-E)}{E-3\mu}$			$\frac{E-2\mu}{2\mu}$	$\frac{\mu E}{3(3\mu-E)}$
$\mu, \nu$	$\frac{2\mu\nu}{1-2\nu}$		$2\nu(1+\nu)$		$\frac{2\mu(1+\nu)}{3(1-2\nu)}$
$\mu, K$	$\frac{3K-2\mu}{3}$		$\frac{9K\mu}{3K+\mu}$	$\frac{3K-2\mu}{2(3K+\mu)}$	
$E, \nu$	$\frac{\nu E}{(1-2\nu)(1+\nu)}$	$\frac{E}{2(1+\nu)}$			$\frac{E}{3(1-2\nu)}$
EK	$\frac{3K(3-E)}{9K-E}$	$\frac{3EK}{9K-E}$		$\frac{3K-E}{6K}$	
$\nu, K$	$\frac{3K\nu}{1+\nu}$	$\frac{3K(1-2\nu)}{2(1+\nu)}$	$3K(1-2\nu)$		

$$A^* = \sqrt{(E-3\lambda)^2 + 8\lambda E}$$

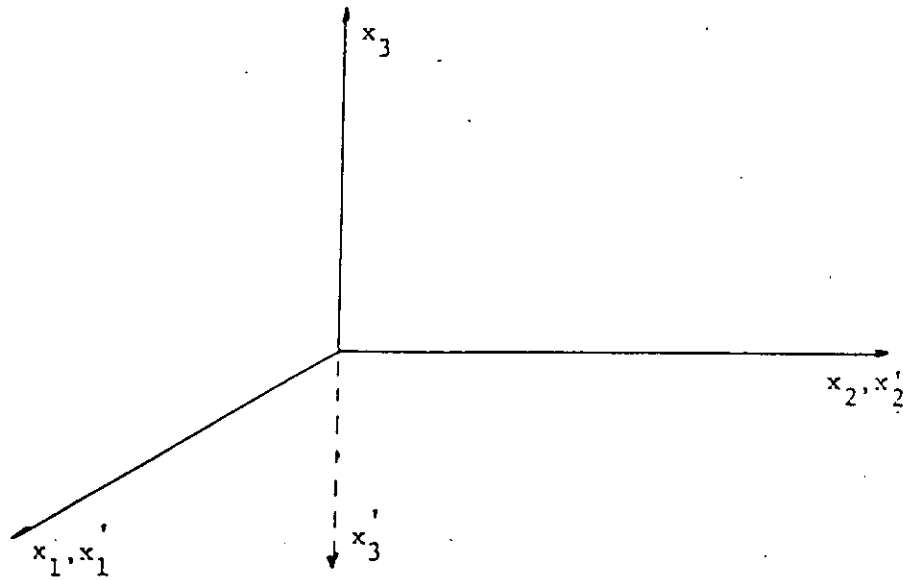


Fig.3.2-1 Monodinic system

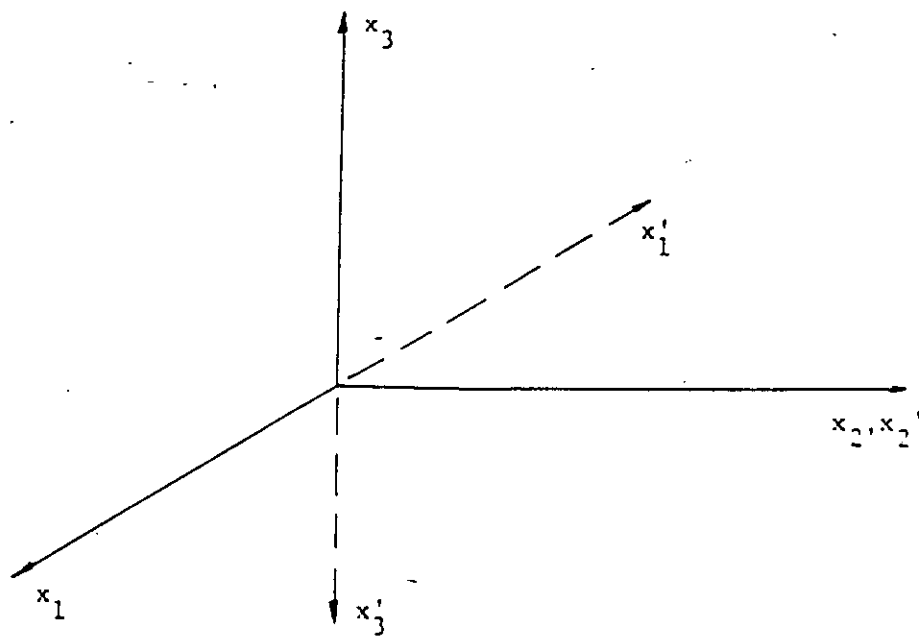


Fig.3.2-2 Orthotropic system

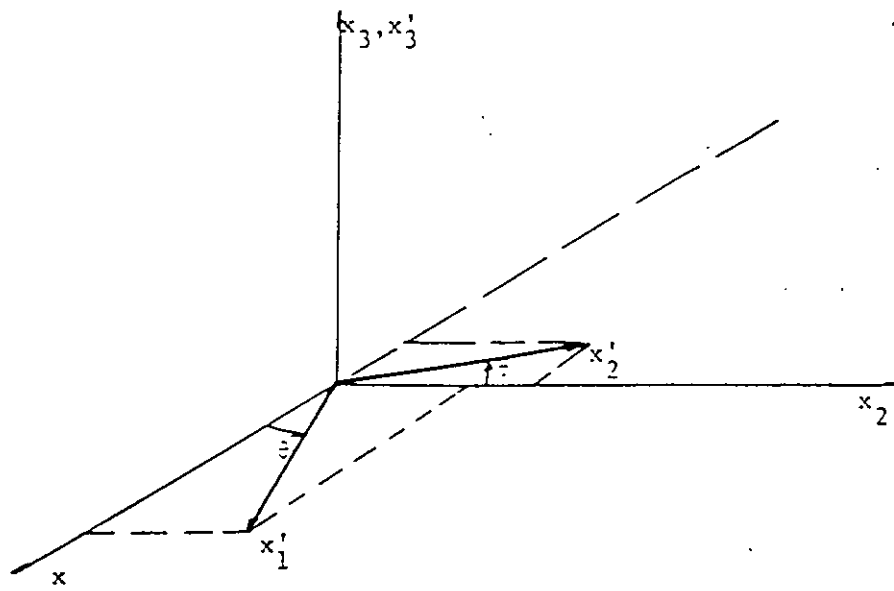


Fig.3.2-3 Transversely isotropic system