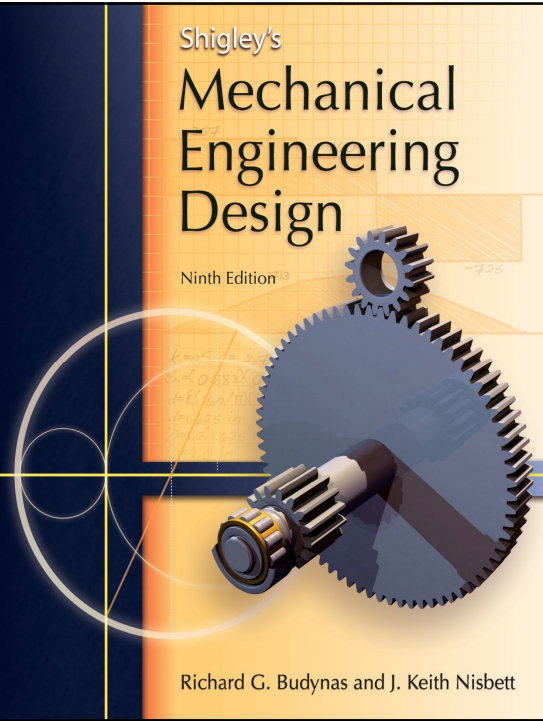


Lecture Slides

Screws, Fasteners,
and the Design of
Nonpermanent Joints

Shigley's
**Mechanical
Engineering
Design**
Ninth Edition¹³



Richard G. Budynas and J. Keith Nisbett

Reasons for Non-permanent Fasteners

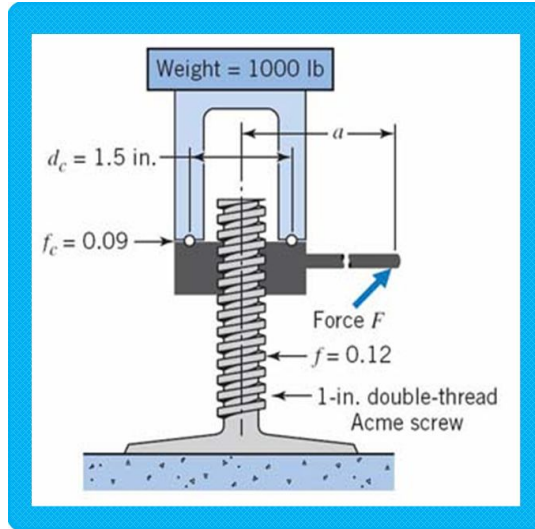
- Field assembly
- Disassembly
- Maintenance
- Adjustment



Introduction

There are two distinct uses for screw threads and:

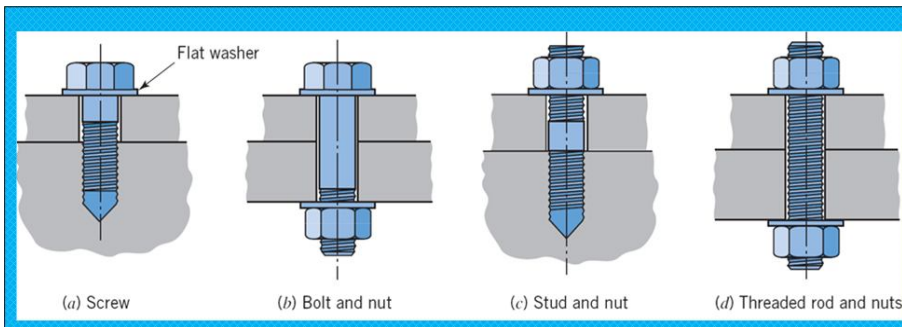
- a **power screw** such as a lathe lead screw or the screw in a car lifting jack which transforms rotary motion into substantial linear motion (or vice versa in certain applications), and



3

Fasteners

Threaded Fastener is similar to what the bolt which joins a number of components together again by transforming rotary motion into linear motion, though in this case the translation is small.



4

Threads

(a) Single (STANDARD)-, (b) double-, and (c) triple threaded screws.

5

Thread Standards and Definitions

- *Pitch* – distance between adjacent threads.
Reciprocal of threads per inch
- *Major diameter* – largest diameter of thread
- *Minor diameter* – smallest diameter of thread
- *Pitch diameter* – theoretical diameter between major and minor diameters, where tooth and gap are same width

Fig. 8-1

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Standardization

- The *American National (Unified)* thread standard defines basic thread geometry for uniformity and interchangeability
- American National (Unified) thread
 - UN normal thread
 - UNR greater root radius for fatigue applications
- Metric thread
 - M series (normal thread)
 - MJ series (greater root radius)

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Standardization

- Coarse series UNC
 - General assembly
 - Frequent disassembly
 - Not good for vibrations
 - The “normal” thread to specify
- Fine series UNF
 - Good for vibrations
 - Good for adjustments
 - Automotive and aircraft
- Extra Fine series UNEF
 - Good for shock and large vibrations
 - High grade alloy
 - Instrumentation
 - Aircraft

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Standardization

- Basic profile for metric M and MJ threads shown in Fig. 8–2
- Tables 8–1 and 8–2 define basic dimensions for standard threads

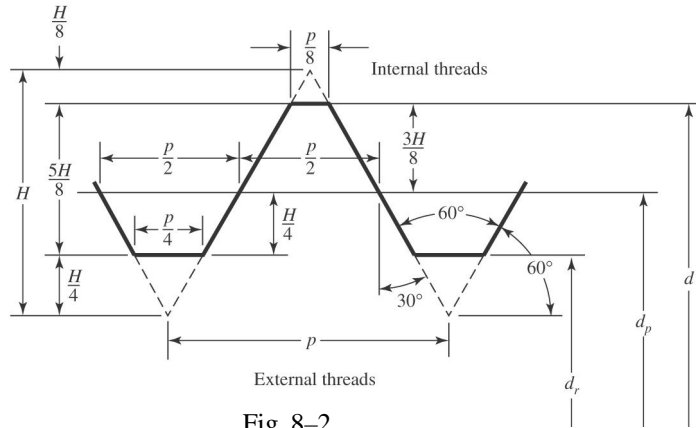


Fig. 8–2

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Diameters and Areas for Metric Threads

Table 8–1

Diameters and Areas of Coarse-Pitch and Fine-Pitch Metric Threads.*

Nominal Major Diameter d mm	Coarse-Pitch Series			Fine-Pitch Series		
	Pitch p mm	Tensile-Stress Area A_t mm^2	Minor-Diameter Area A_r mm^2	Pitch p mm	Tensile-Stress Area A_t mm^2	Minor-Diameter Area A_r mm^2
1.6	0.35	1.27	1.07			
2	0.40	2.07	1.79			
2.5	0.45	3.39	2.98			
3	0.5	5.03	4.47			
3.5	0.6	6.78	6.00			
4	0.7	8.78	7.75			
5	0.8	14.2	12.7			
6	1	20.1	17.9			
8	1.25	36.6	32.8	1	39.2	36.0
10	1.5	58.0	52.3	1.25	61.2	56.3
12	1.75	84.3	76.3	1.25	92.1	86.0
14	2	115	104	1.5	125	116
16	2	157	144	1.5	167	157
20	2.5	245	225	1.5	272	259
24	3	353	324	2	384	365
30	3.5	561	519	2	621	596
36	4	817	759	2	915	884
42	4.5	1120	1050	2	1260	1230
48	5	1470	1380	2	1670	1630
56	5.5	2030	1910	2	2300	2250
64	6	2680	2520	2	3030	2980

Diameters and Areas for Unified Screw Threads

Table 8-2

Size Designation	Nominal Major Diameter in	Coarse Series—UNC			Fine Series—UNF		
		Threads per Inch N	Tensile-Stress Area A_t , in ²	Minor-Diameter Area A_r , in ²	Threads per Inch N	Tensile-Stress Area A_t , in ²	Minor-Diameter Area A_r , in ²
0	0.0600				80	0.001 80	0.001 51
1	0.0730	64	0.002 63	0.002 18	72	0.002 78	0.002 37
2	0.0860	56	0.003 70	0.003 10	64	0.003 94	0.003 39
3	0.0990	48	0.004 87	0.004 06	56	0.005 23	0.004 51
4	0.1120	40	0.006 04	0.004 96	48	0.006 61	0.005 66
5	0.1250	40	0.007 96	0.006 72	44	0.008 80	0.007 16
6	0.1380	32	0.009 09	0.007 45	40	0.010 15	0.008 74
8	0.1640	32	0.014 0	0.011 96	36	0.014 74	0.012 85
10	0.1900	24	0.017 5	0.014 50	32	0.020 0	0.017 5
12	0.2160	24	0.024 2	0.020 6	28	0.025 8	0.022 6
$\frac{1}{4}$	0.2500	20	0.031 8	0.026 9	28	0.036 4	0.032 6
$\frac{5}{16}$	0.3125	18	0.052 4	0.045 4	24	0.058 0	0.052 4
$\frac{3}{8}$	0.3750	16	0.077 5	0.067 8	24	0.087 8	0.080 9
$\frac{7}{16}$	0.4375	14	0.106 3	0.093 3	20	0.118 7	0.109 0
$\frac{1}{2}$	0.5000	13	0.141 9	0.125 7	20	0.159 9	0.148 6
$\frac{9}{16}$	0.5625	12	0.182	0.162	18	0.203	0.189
$\frac{5}{8}$	0.6250	11	0.226	0.202	18	0.256	0.240
$\frac{3}{4}$	0.7500	10	0.334	0.302	16	0.373	0.351
$\frac{7}{8}$	0.8750	9	0.462	0.419	14	0.509	0.480
1	1.0000	8	0.606	0.551	12	0.663	0.625
$1\frac{1}{4}$	1.2500	7	0.969	0.890	12	1.073	1.024
$1\frac{1}{2}$	1.5000	6	1.405	1.294	12	1.581	1.521

Tensile Stress Area

- The tensile stress area, A_t , is the area of an unthreaded rod with the same tensile strength as a threaded rod.
- It is the effective area of a threaded rod to be used for stress calculations.
- The diameter of this unthreaded rod is the average of the pitch diameter and the minor diameter of the threaded rod.

Square and Acme Threads

- Square and Acme threads are used when the threads are intended to transmit power

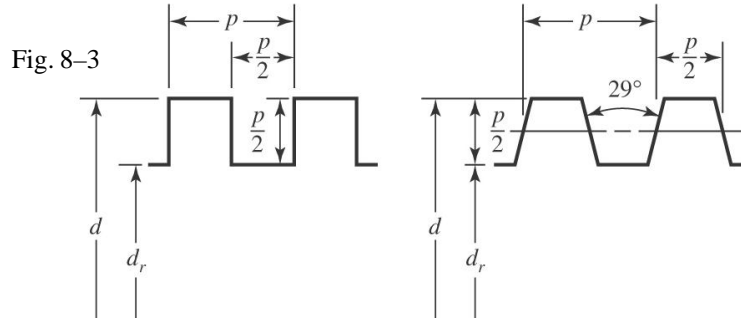


Table 8-3 Preferred Pitches for Acme Threads

$d, \text{ in}$	$\frac{1}{4}$	$\frac{5}{16}$	$\frac{3}{8}$	$\frac{1}{2}$	$\frac{5}{8}$	$\frac{3}{4}$	$\frac{7}{8}$	1	$1\frac{1}{4}$	$1\frac{1}{2}$	$1\frac{3}{4}$	2	$2\frac{1}{2}$	3
$p, \text{ in}$	$\frac{1}{16}$	$\frac{1}{14}$	$\frac{1}{12}$	$\frac{1}{10}$	$\frac{1}{8}$	$\frac{1}{6}$	$\frac{1}{6}$	$\frac{1}{5}$	$\frac{1}{5}$	$\frac{1}{4}$	$\frac{1}{4}$	$\frac{1}{4}$	$\frac{1}{3}$	$\frac{1}{2}$

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Mechanics of Power Screws

- *Power screw*
 - Used to change angular motion into linear motion
 - Usually transmits power
 - Examples include vises, presses, jacks, lead screw on lathe

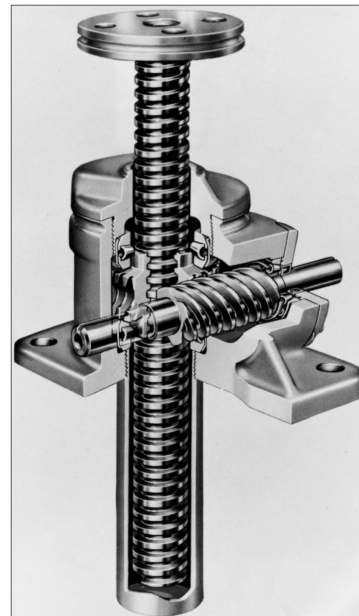
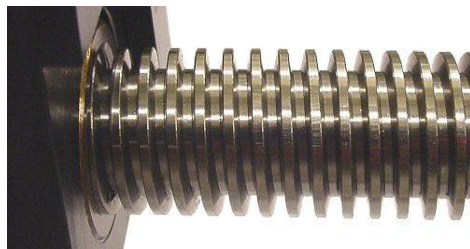


Fig. 8-4

Shigley's Mechanical Engineering Design

Mechanics of Power Screws

(a) (b) (c)

Weight supported by three screw jacks. In each screw jack, only the shaded member rotates.

CH-8 LEC 33 Slide 15

Mechanics of Power Screws

- Find expression for torque required to raise or lower a load
- Unroll one turn of a thread
- Treat thread as inclined plane
- Do force analysis

Fig. 8-5

(a) (b)

Fig. 8-6

Shigley's Mechanical Engineering Design

Mechanics of Power Screws

- For raising the load

$$\sum F_x = P_R - N \sin \lambda - fN \cos \lambda = 0 \tag{a}$$

$$\sum F_y = -F - fN \sin \lambda + N \cos \lambda = 0$$

- For lowering the load

$$\sum F_x = -P_L - N \sin \lambda + fN \cos \lambda = 0 \tag{b}$$

$$\sum F_y = -F + fN \sin \lambda + N \cos \lambda = 0$$

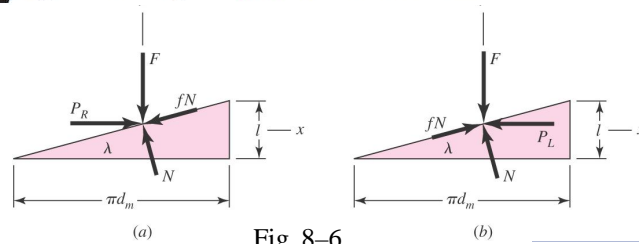


Fig. 8-6

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Mechanics of Power Screws

- Eliminate N and solve for P to raise and lower the load

$$P_R = \frac{F(\sin \lambda + f \cos \lambda)}{\cos \lambda - f \sin \lambda} \tag{c}$$

$$P_L = \frac{F(f \cos \lambda - \sin \lambda)}{\cos \lambda + f \sin \lambda} \tag{d}$$

- Divide numerator and denominator by $\cos l$ and use relation $\tan l = l/\rho d_m$

$$P_R = \frac{F[(l/\pi d_m) + f]}{1 - (fl/\pi d_m)} \tag{e}$$

$$P_L = \frac{F[f - (l/\pi d_m)]}{1 + (fl/\pi d_m)} \tag{f}$$

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Raising and Lowering Torque

- Noting that the torque is the product of the force and the mean radius,

$$T_R = \frac{F d_m}{2} \left(\frac{l + \pi f d_m}{\pi d_m - f l} \right) \quad (8-1)$$

$$T_L = \frac{F d_m}{2} \left(\frac{\pi f d_m - l}{\pi d_m + f l} \right) \quad (8-2)$$

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Self-locking Condition

$$T_L = \frac{F d_m}{2} \left(\frac{\pi f d_m - l}{\pi d_m + f l} \right) \quad (8-2)$$

- If the lowering torque is negative, the load will lower itself by causing the screw to spin without any external effort.
- If the lowering torque is positive, the screw is *self-locking*.
- Self-locking condition is $\rho f d_m > l$
- Noting that $l / \rho d_m = \tan l$, the self-locking condition can be seen to only involve the coefficient of friction and the lead angle.

$$f > \tan \lambda \quad (8-3)$$

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Power Screw Efficiency

- The torque needed to raise the load with no friction losses can be found from Eq. (8-1) with $f = 0$.

$$T_0 = \frac{Fl}{2\pi} \tag{g}$$

- The efficiency of the power screw is therefore

$$e = \frac{T_0}{T_R} = \frac{Fl}{2\pi T_R} \tag{8-4}$$

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Power Screws with Acme Threads

- If Acme threads are used instead of square threads, the thread angle creates a wedging action.
- The friction components are increased.
- The torque necessary to raise a load (or tighten a screw) is found by dividing the friction terms in Eq. (8-1) by $\cos\alpha$.

$$T_R = \frac{F d_m}{2} \left(\frac{l + \pi f d_m \sec \alpha}{\pi d_m - f l \sec \alpha} \right) \tag{8-5}$$

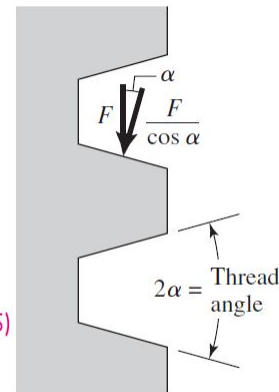


Fig. 8-7 (a)

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Collar Friction

- An additional component of torque is often needed to account for the friction between a collar and the load.
- Assuming the load is concentrated at the mean collar diameter d_c

$$T_c = \frac{F f_c d_c}{2} \quad (8-6)$$

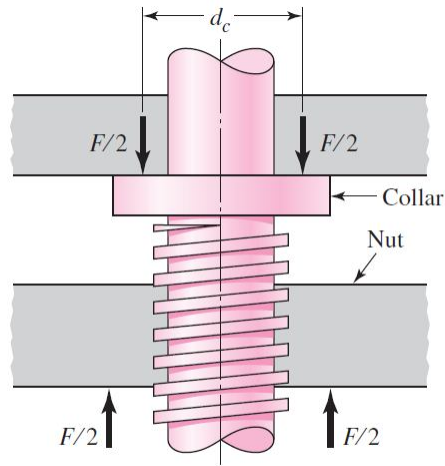


Fig. 8-7 (b)

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Stresses in Body of Power Screws

- Maximum nominal shear stress in torsion of the screw body

$$\tau = \frac{16T}{\pi d_r^3} \quad (8-7)$$

- Axial stress in screw body

$$\sigma = \frac{F}{A} = \frac{4F}{\pi d_r^2} \quad (8-8)$$

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Stresses in Threads of Power Screws

- Bearing stress in threads,

$$\sigma_B = -\frac{F}{\pi d_m n_t p/2} = -\frac{2F}{\pi d_m n_t p} \quad (8-10)$$

where n_t is number of engaged threads

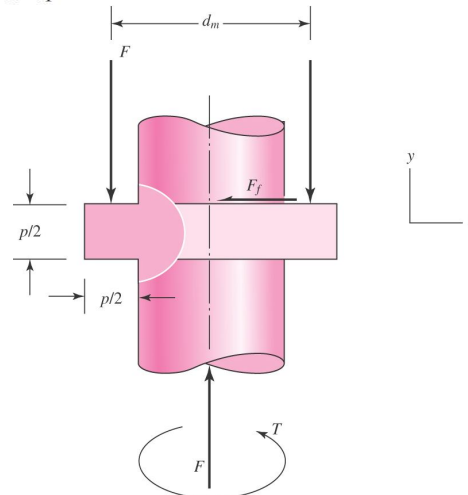


Fig. 8-8

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Stresses in Threads of Power Screws

- Bending stress at root of thread,

$$Z = \frac{I}{c} = \frac{(\pi d_r n_t) (p/2)^2}{6} = \frac{\pi}{24} d_r n_t p^2$$

$$M = \frac{F p}{4}$$

$$\sigma_b = \frac{M}{Z} = \frac{F p}{4} \frac{24}{\pi d_r n_t p^2} = \frac{6F}{\pi d_r n_t p} \quad (8-11)$$

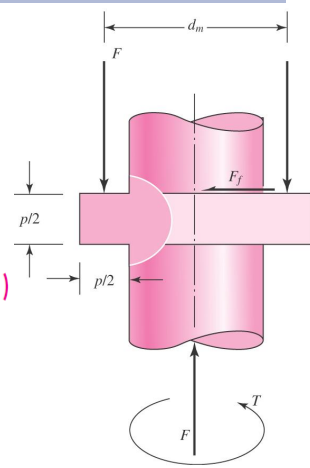


Fig. 8-8

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Stresses in Threads of Power Screws

- Transverse shear stress at center of root of thread,

$$\tau = \frac{3V}{2A} = \frac{3}{2} \frac{F}{\pi d_r n_t p/2} = \frac{3F}{\pi d_r n_t p}$$

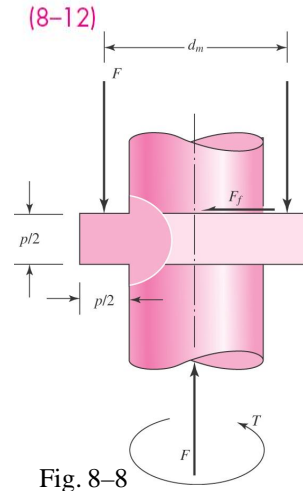


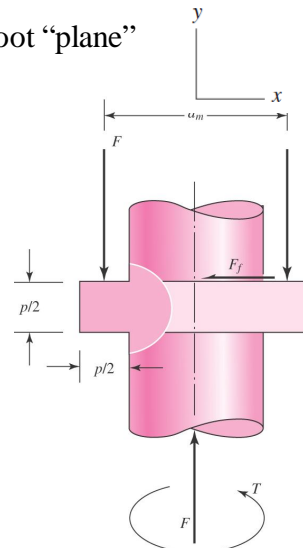
Fig. 8-8

Shigley's Mechanical Engineering Design

Stresses in Threads of Power Screws

- Consider stress element at the top of the root "plane"

$$\begin{aligned} \sigma_x &= \frac{6F}{\pi d_r n_t p} & \tau_{xy} &= 0 \\ \sigma_y &= -\frac{4F}{\pi d_r^2} & \tau_{yz} &= \frac{16T}{\pi d_r^3} \\ \sigma_z &= 0 & \tau_{zx} &= 0 \end{aligned}$$



- Obtain von Mises stress from Eq. (5-14),

$$\sigma' = \frac{1}{\sqrt{2}} [(\sigma_x - \sigma_y)^2 + (\sigma_y - \sigma_z)^2 + (\sigma_z - \sigma_x)^2 + 6(\tau_{xy}^2 + \tau_{yz}^2 + \tau_{zx}^2)]^{1/2} \quad (5-14)$$

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Thread Deformation in Screw-Nut Combination

- Power screw thread is in compression, causing elastic shortening of screw thread pitch.
- Engaging nut is in tension, causing elastic lengthening of the nut thread pitch.
- Consequently, the engaged threads cannot share the load equally.
- Experiments indicate the first thread carries 38% of the load, the second thread 25%, and the third thread 18%. The seventh thread is free of load.
- To find the largest stress in the first thread of a screw-nut combination, use $0.38F$ in place of F , and set $n_t = 1$.

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Power Screw Safe Bearing Pressure

Table 8-4

Screw Bearing

Pressure p_b

Source: H. A. Rothbart and T. H. Brown, Jr., *Mechanical Design Handbook*, 2nd ed., McGraw-Hill, New York, 2006.

Screw Material	Nut Material	Safe p_b , psi	Notes
Steel	Bronze	2500–3500	Low speed
Steel	Bronze	1600–2500	≤ 10 fpm
	Cast iron	1800–2500	≤ 8 fpm
Steel	Bronze	800–1400	20–40 fpm
	Cast iron	600–1000	20–40 fpm
Steel	Bronze	150–240	≥ 50 fpm

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Power Screw Friction Coefficients

Table 8-5

Coefficients of Friction f for Threaded Pairs

Source: H. A. Rothbart and T. H. Brown, Jr., *Mechanical Design Handbook*, 2nd ed., McGraw-Hill, New York, 2006.

Screw Material	Nut Material			
	Steel	Bronze	Brass	Cast Iron
Steel, dry	0.15–0.25	0.15–0.23	0.15–0.19	0.15–0.25
Steel, machine oil	0.11–0.17	0.10–0.16	0.10–0.15	0.11–0.17
Bronze	0.08–0.12	0.04–0.06	—	0.06–0.09

Table 8-6

Thrust-Collar Friction Coefficients

Source: H. A. Rothbart and T. H. Brown, Jr., *Mechanical Design Handbook*, 2nd ed., McGraw-Hill, New York, 2006.

Combination	Running	Starting
Soft steel on cast iron	0.12	0.17
Hard steel on cast iron	0.09	0.15
Soft steel on bronze	0.08	0.10
Hard steel on bronze	0.06	0.08

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Head Type of Bolts

- Hexagon head bolt
 - Usually uses nut
 - Heavy duty
- Hexagon head cap screw
 - Thinner head
 - Often used as screw (in threaded hole, without nut)
- Socket head cap screw
 - Usually more precision applications
 - Access from the top
- Machine screws
 - Usually smaller sizes
 - Slot or philips head common
 - Threaded all the way

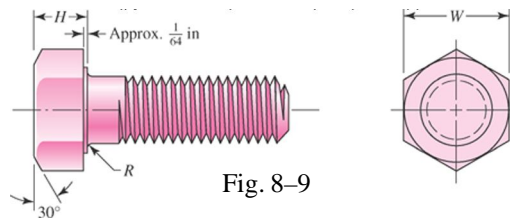


Fig. 8-9

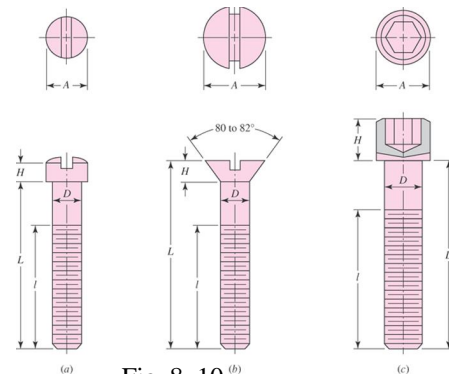
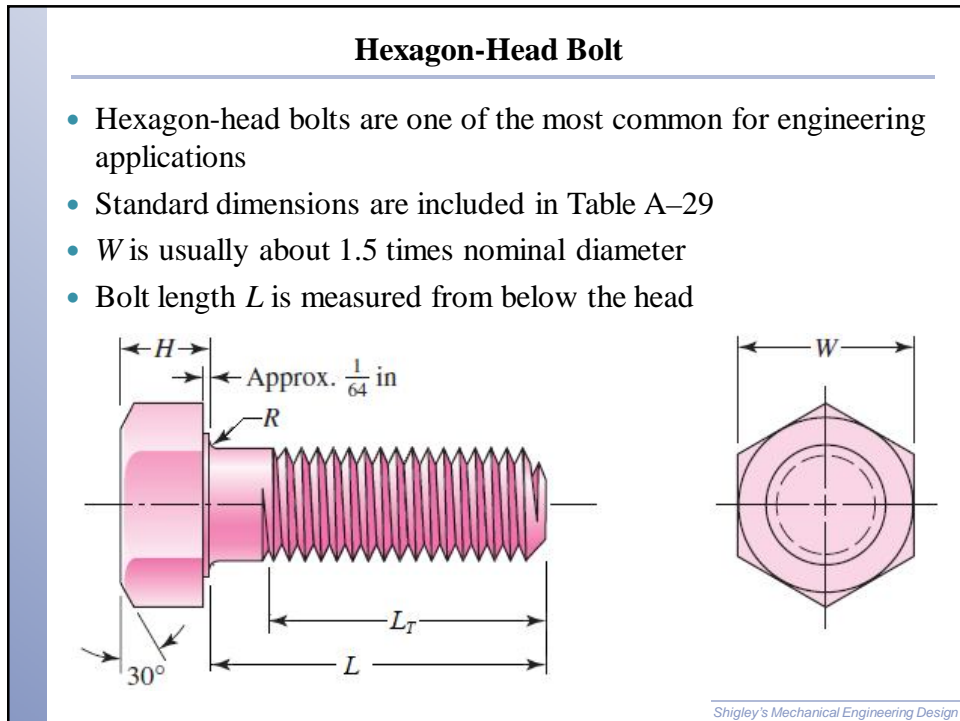
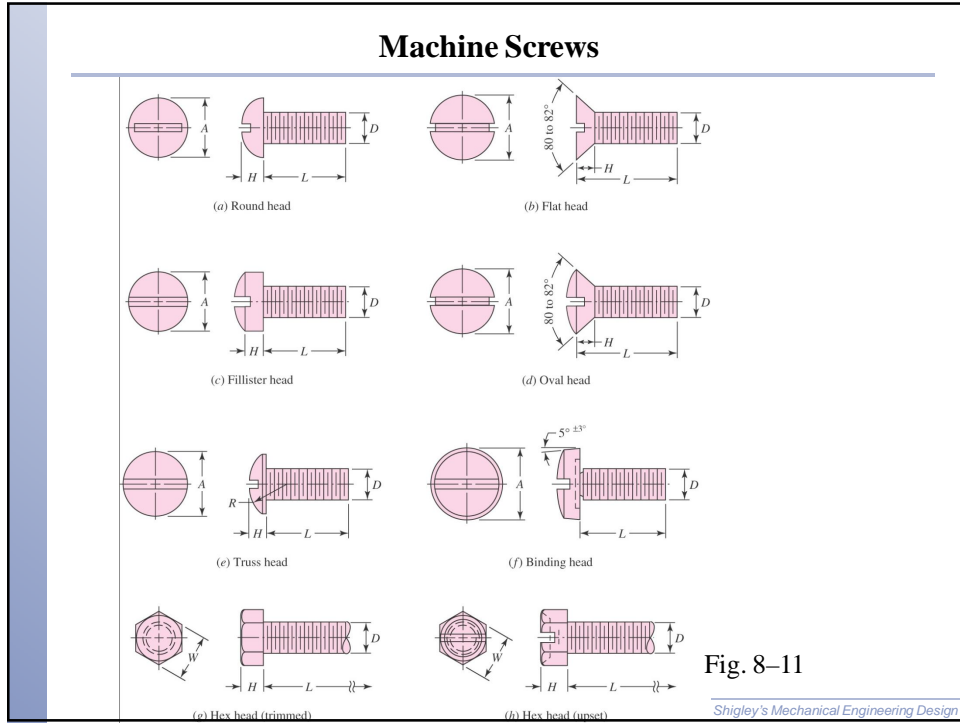


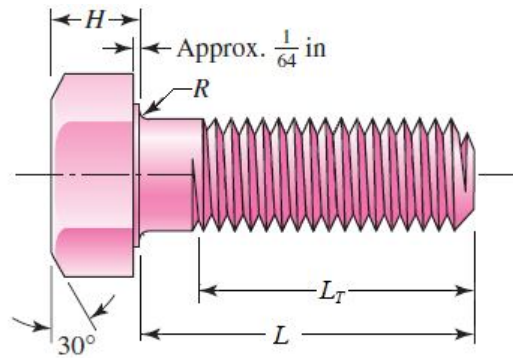
Fig. 8-10 Shigley's Mechanical Engineering Design



Threaded Lengths

English $L_T = \begin{cases} 2d + \frac{1}{4} \text{ in} & L \leq 6 \text{ in} \\ 2d + \frac{1}{2} \text{ in} & L > 6 \text{ in} \end{cases} \quad (8-13)$

Metric $L_T = \begin{cases} 2d + 6 & L \leq 125 & d \leq 48 \\ 2d + 12 & 125 < L \leq 200 \\ 2d + 25 & L > 200 \end{cases} \quad (8-14)$



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Nuts

- See Appendix A-31 for typical specifications
- First three threads of nut carry majority of load
- Localized plastic strain in the first thread is likely, so nuts should not be re-used in critical applications.

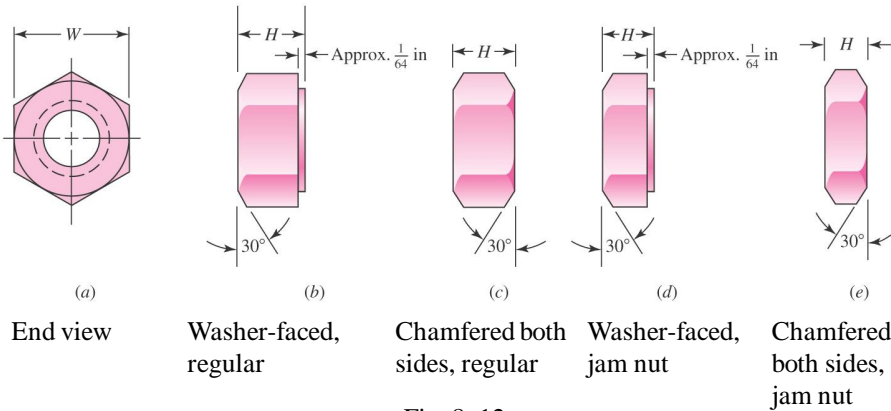


Fig. 8-12

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Tension Loaded Bolted Joint

- Grip length l includes everything being compressed by bolt preload, including washers
- Washer under head prevents burrs at the hole from gouging into the fillet under the bolt head

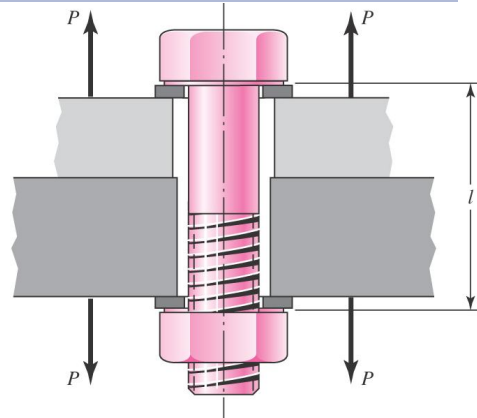


Fig. 8-13

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Pressure Vessel Head

- Hex-head cap screw in tapped hole used to fasten cylinder head to cylinder body
- Note O-ring seal, not affecting the stiffness of the members within the grip
- Only part of the threaded length of the bolt contributes to the effective grip l

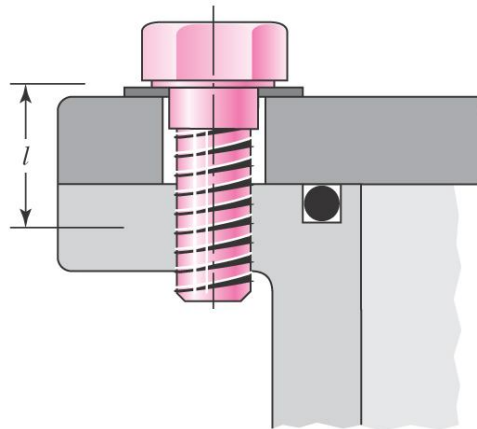


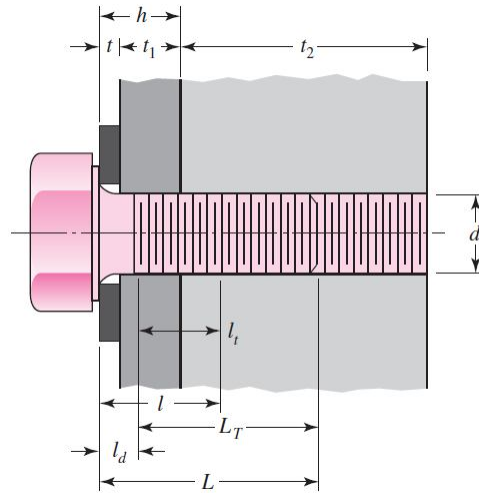
Fig. 8-14

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Effective Grip Length for Tapped Holes

- For screw in tapped hole, effective grip length is

$$l = \begin{cases} h + t_2/2, & t_2 < d \\ h + d/2, & t_2 \geq d \end{cases}$$



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Bolted Joint Stiffnesses

- During bolt preload
 - bolt is stretched
 - members in grip are compressed
- When external load P is applied
 - Bolt stretches further
 - Members in grip uncompress some
- Joint can be modeled as a soft bolt spring in parallel with a stiff member spring

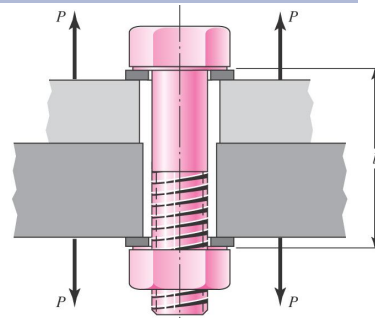
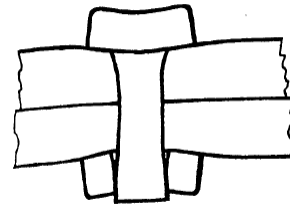


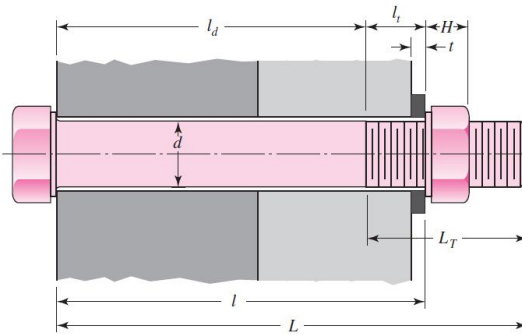
Fig. 8-13



Shigley's Mechanical Engineering Design

Bolt Stiffness

- Axially loaded rod, partly threaded and partly unthreaded
- Consider each portion as a spring
- Combine as two springs in series



$$\frac{1}{k} = \frac{1}{k_1} + \frac{1}{k_2} \quad \text{or} \quad k = \frac{k_1 k_2}{k_1 + k_2} \quad (8-15)$$

$$k_t = \frac{A_t E}{l_t} \quad k_d = \frac{A_d E}{l_d} \quad (8-16)$$

$$k_b = \frac{A_d A_t E}{A_d l_t + A_t l_d} \quad (8-17)$$

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Procedure to Find Bolt Stiffness

Given fastener diameter d and pitch p in mm or number of threads per inch

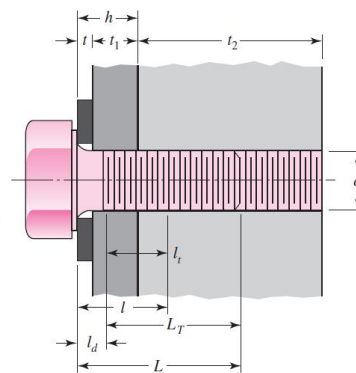
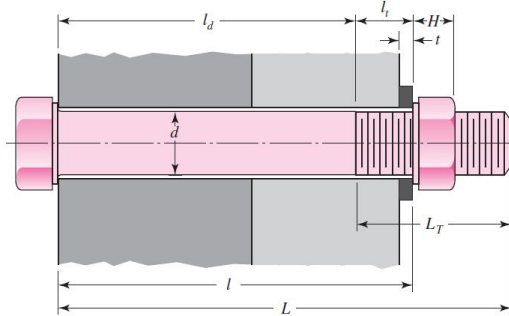
Washer thickness: t from Table A-32 or A-33

Nut thickness [Fig. (a) only]: H from Table A-31

Grip length:

For Fig. (a): $l =$ thickness of all material squeezed between face of bolt and face of nut

$$\text{For Fig. (b): } l = \begin{cases} h + t_2/2, & t_2 < d \\ h + d/2, & t_2 \geq d \end{cases}$$



Shigley's Mechanical Engineering Design

Procedure to Find Bolt Stiffness

Fastener length (round up using Table A-17*):

For Fig. (a): $L > l + H$

For Fig. (b): $L > h + 1.5d$

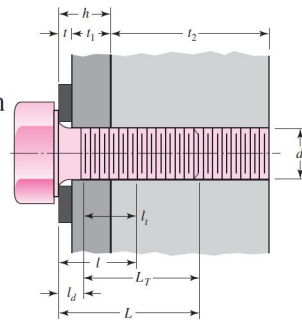
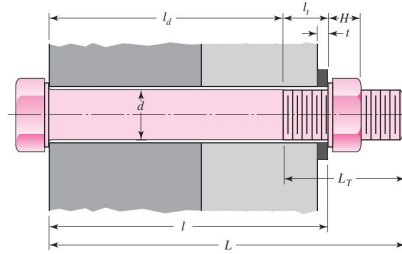
Threaded length L_T :

Inch series:

$$L_T = \begin{cases} 2d + \frac{1}{4} \text{ in}, & L \leq 6 \text{ in} \\ 2d + \frac{1}{2} \text{ in}, & L > 6 \text{ in} \end{cases}$$

Metric series:

$$L_T = \begin{cases} 2d + 6 \text{ mm}, & L \leq 125 \text{ mm}, d \leq 48 \text{ mm} \\ 2d + 12 \text{ mm}, & 125 < L \leq 200 \text{ mm} \\ 2d + 25 \text{ mm}, & L > 200 \text{ mm} \end{cases}$$



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Procedure to Find Bolt Stiffness

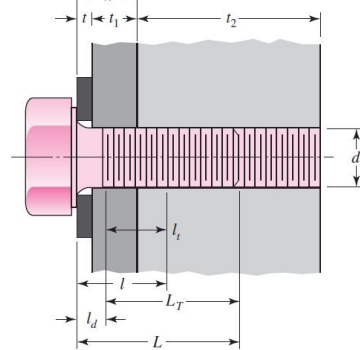
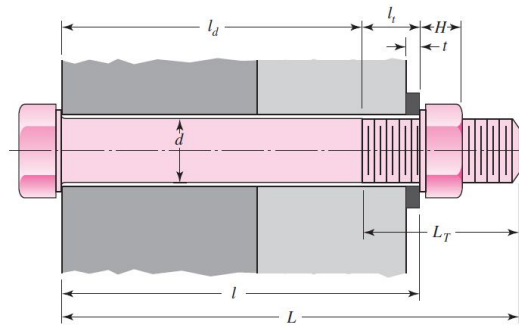
Length of unthreaded portion in grip: $l_d = L - L_T$

Length of threaded portion in grip: $l_t = l - l_d$

Area of unthreaded portion: $A_d = \pi d^2/4$

Area of threaded portion: A_t from Table 8-1 or 8-2

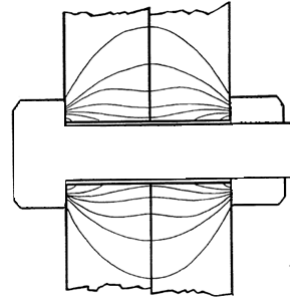
Fastener stiffness: $k_b = \frac{A_d A_t E}{A_d l_t + A_t l_d}$



Shigley's Mechanical Engineering Design

Member Stiffness

- Stress distribution spreads from face of bolt head and nut
- Model as a cone with top cut off
- Called a *frustum*



Shigley's Mechanical Engineering Design

Member Stiffness

- Model compressed members as if they are frusta spreading from the bolt head and nut to the midpoint of the grip
- Each frustum has a half-apex angle of α
- Find stiffness for frustum in compression

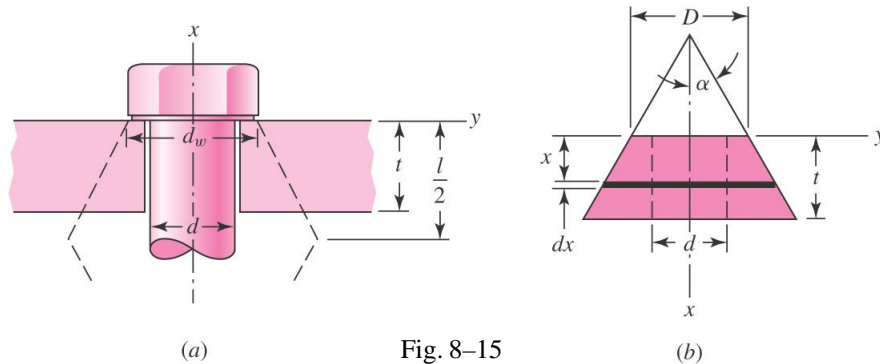


Fig. 8-15

Shigley's Mechanical Engineering Design

Member Stiffness

$$d\delta = \frac{P dx}{EA} \quad (a)$$

$$A = \pi(r_o^2 - r_i^2) = \pi \left[\left(x \tan \alpha + \frac{D}{2} \right)^2 - \left(\frac{d}{2} \right)^2 \right] \quad (b)$$

$$= \pi \left(x \tan \alpha + \frac{D+d}{2} \right) \left(x \tan \alpha + \frac{D-d}{2} \right)$$

$$\delta = \frac{P}{\pi E} \int_0^t \frac{dx}{[x \tan \alpha + (D+d)/2][x \tan \alpha + (D-d)/2]} \quad (c)$$

$$\delta = \frac{P}{\pi E d \tan \alpha} \ln \frac{(2t \tan \alpha + D - d)(D + d)}{(2t \tan \alpha + D + d)(D - d)} \quad (d)$$

$$k = \frac{P}{\delta} = \frac{\pi E d \tan \alpha}{\ln \frac{(2t \tan \alpha + D - d)(D + d)}{(2t \tan \alpha + D + d)(D - d)}} \quad (8-19)$$

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Member Stiffness

- With typical value of $\alpha = 30^\circ$,

$$k = \frac{0.5774 \pi E d}{\ln \frac{(1.155t + D - d)(D + d)}{(1.155t + D + d)(D - d)}} \quad (8-20)$$

- Use Eq. (8-20) to find stiffness for each frustum
- Combine all frusta as springs in series

$$\frac{1}{k_m} = \frac{1}{k_1} + \frac{1}{k_2} + \frac{1}{k_3} + \dots + \frac{1}{k_i} \quad (8-18)$$

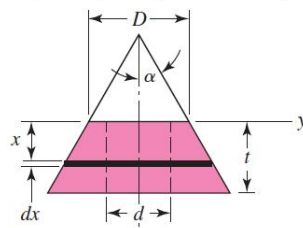


Fig. 8-15b

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Member Stiffness for Common Material in Grip

- If the grip consists of any number of members all of the same material, two identical frusta can be added in series. The entire joint can be handled with one equation,

$$k_m = \frac{\pi E d \tan \alpha}{2 \ln \frac{(l \tan \alpha + d_w - d)(d_w + d)}{(l \tan \alpha + d_w + d)(d_w - d)}} \quad (8-21)$$

- d_w is the washer face diameter
- Using standard washer face diameter of $1.5d$, and with $a = 30^\circ$,

$$k_m = \frac{0.5774\pi E d}{2 \ln \left(5 \frac{0.5774l + 0.5d}{0.5774l + 2.5d} \right)} \quad (8-22)$$

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Bolt Materials

- Grades specify material, heat treatment, strengths
 - Table 8–9 for SAE grades
 - Table 8–10 for ASTM designations
 - Table 8–11 for metric property class
- Grades should be marked on head of bolt

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Bolt Materials

- *Proof load* is the maximum load that a bolt can withstand without acquiring a permanent set
- *Proof strength* is the quotient of proof load and tensile-stress area
 - Corresponds to proportional limit
 - Slightly lower than yield strength
 - Typically used for static strength of bolt
- Good bolt materials have stress-strain curve that continues to rise to fracture

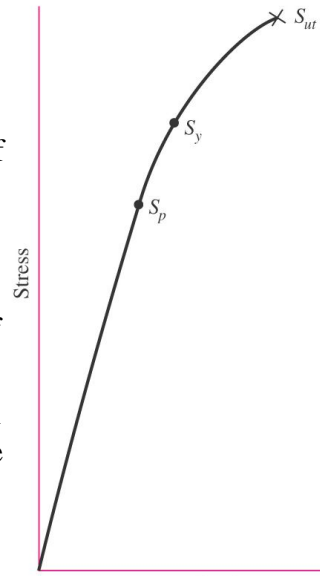


Fig. 8-18 Strain

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SAE Specifications for Steel Bolts

Table 8-9

SAE Grade No.	Size Range Inclusive, in	Minimum Proof Strength,* kpsi	Minimum Tensile Strength,* kpsi	Minimum Yield Strength,* kpsi	Material	Head Marking
1	$\frac{1}{4}$ - $1\frac{1}{2}$	33	60	36	Low or medium carbon	
2	$\frac{1}{4}$ - $\frac{3}{8}$ $\frac{7}{8}$ - $1\frac{1}{2}$	55	74	57	Low or medium carbon	
		33	60	36		
4	$\frac{1}{4}$ - $1\frac{1}{2}$	65	115	100	Medium carbon, cold-drawn	
5	$\frac{1}{4}$ -1 $1\frac{1}{8}$ - $1\frac{1}{2}$	85	120	92	Medium carbon, Q&T	
		74	105	81		
5.2	$\frac{1}{4}$ -1	85	120	92	Low-carbon martensite, Q&T	
7	$\frac{1}{4}$ - $1\frac{1}{2}$	105	133	115	Medium-carbon alloy, Q&T	
8	$\frac{1}{4}$ - $1\frac{1}{2}$	120	150	130	Medium-carbon alloy, Q&T	
8.2	$\frac{1}{4}$ -1	120	150	130	Low-carbon martensite, Q&T	

*Minimum strengths are strengths exceeded by 99 percent of fasteners.

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ASTM Specification for Steel Bolts

Table 8-10

ASTM Designation No.	Size Range, Inclusive, in	Minimum Proof Strength, [†] kpsi	Minimum Tensile Strength, [†] kpsi	Minimum Yield Strength, [†] kpsi	Material	Head Marking
A307	$\frac{1}{4}$ - $1\frac{1}{2}$	33	60	36	Low carbon	
A325, type 1	$\frac{1}{2}$ -1	85	120	92	Medium carbon, Q&T	
	$1\frac{1}{4}$ - $1\frac{1}{2}$	74	105	81		
A325, type 2	$\frac{1}{2}$ -1	85	120	92	Low-carbon, martensite, Q&T	
	$1\frac{1}{4}$ - $1\frac{1}{2}$	74	105	81		
A325, type 3	$\frac{1}{2}$ -1	85	120	92	Weathering steel, Q&T	
	$1\frac{1}{4}$ - $1\frac{1}{2}$	74	105	81		
A354, grade BC	$\frac{1}{2}$ - $2\frac{1}{2}$	105	125	109	Alloy steel, Q&T	
	$2\frac{1}{2}$ -4	95	115	99		
A354, grade BD	$\frac{1}{2}$ -4	120	150	130	Alloy steel, Q&T	
A449	$\frac{1}{2}$ -1	85	120	92	Medium-carbon, Q&T	
	$1\frac{1}{8}$ - $1\frac{1}{2}$	74	105	81		
	$1\frac{1}{2}$ -3	55	90	58		
A490, type 1	$\frac{1}{2}$ - $1\frac{1}{2}$	120	150	130	Alloy steel, Q&T	
A490, type 3	$\frac{1}{2}$ - $1\frac{1}{2}$	120	150	130	Weathering steel, Q&T	

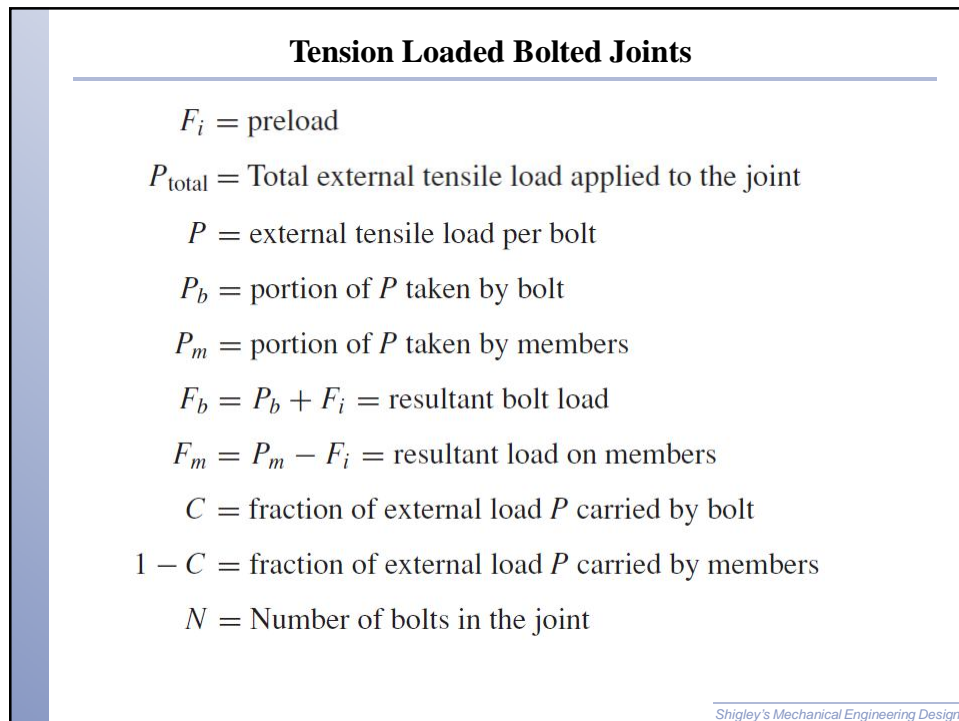
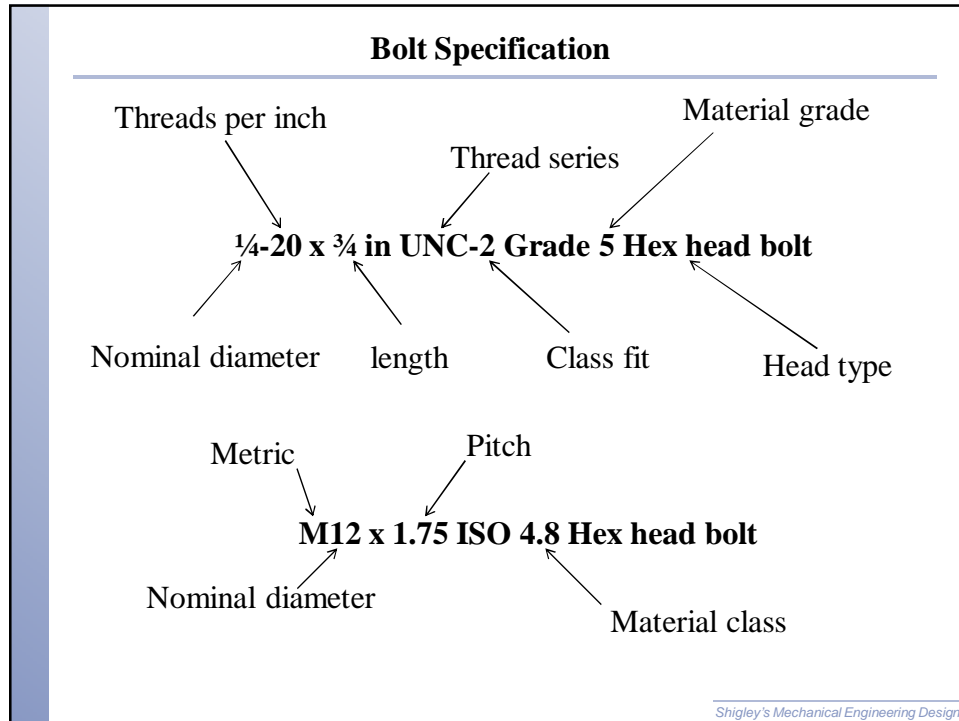
[†]Minimum strengths are strengths exceeded by 99 percent of fasteners.

Metric Mechanical-Property Classes for Steel Bolts

Property Class	Size Range, Inclusive	Minimum Proof Strength, [†] MPa	Minimum Tensile Strength, [†] MPa	Minimum Yield Strength, [†] MPa	Material	Head Marking
4.6	M5-M36	225	400	240	Low or medium carbon	
4.8	M1.6-M16	310	420	340	Low or medium carbon	
5.8	M5-M24	380	520	420	Low or medium carbon	
8.8	M16-M36	600	830	660	Medium carbon, Q&T	
9.8	M1.6-M16	650	900	720	Medium carbon, Q&T	
10.9	M5-M36	830	1040	940	Low-carbon martensite, Q&T	
12.9	M1.6-M36	970	1220	1100	Alloy, Q&T	

Table 8-11

[†]The thread length for bolts and cap screws is



Tension Loaded Bolted Joints

- During bolt preload
 - bolt is stretched
 - members in grip are compressed
- When external load P is applied
 - Bolt stretches an additional amount δ
 - Members in grip uncompress same amount δ

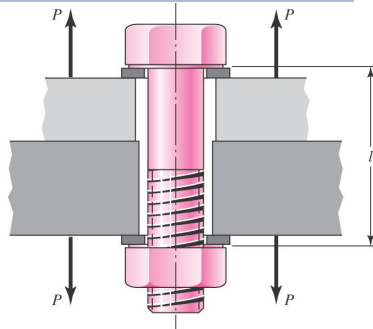


Fig. 8-13

$$\delta = \frac{P_b}{k_b} \quad \text{and} \quad \delta = \frac{P_m}{k_m} \tag{b}$$

$$P_m = P_b \frac{k_m}{k_b} \tag{c}$$

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Stiffness Constant

- Since $P = P_b + P_m$

$$P_b = \frac{k_b P}{k_b + k_m} = C P \tag{d}$$

$$P_m = P - P_b = (1 - C) P \tag{e}$$

- C is defined as the *stiffness constant* of the joint

$$C = \frac{k_b}{k_b + k_m} \tag{f}$$

- C indicates the proportion of external load P that the bolt will carry. A good design target is around 0.2.

Table 8-12

Computation of Bolt and Member Stiffnesses. Steel members clamped using a $\frac{1}{2}$ in-13 NC steel bolt. $C = \frac{k_b}{k_b + k_m}$

Bolt Grip, in	Stiffnesses, M lbf/in			
	k_b	k_m	C	$1 - C$
2	2.57	12.69	0.168	0.832
3	1.79	11.33	0.136	0.864
4	1.37	10.63	0.114	0.886

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Bolt and Member Loads

- The resultant bolt load is

$$F_b = P_b + F_i = C P + F_i \quad F_m < 0 \quad (8-24)$$

- The resultant load on the members is

$$F_m = P_m - F_i = (1 - C)P - F_i \quad F_m < 0 \quad (8-25)$$

- These results are only valid if the load on the members remains negative, indicating the members stay in compression.

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Relating Bolt Torque to Bolt Tension

- Best way to measure bolt preload is by relating measured bolt elongation and calculated stiffness
- Usually, measuring bolt elongation is not practical
- Measuring applied torque is common, using a torque wrench
- Need to find relation between applied torque and bolt preload

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Relating Bolt Torque to Bolt Tension

- From the power screw equations, Eqs. (8-5) and (8-6), we get

$$T = \frac{F_i d_m}{2} \left(\frac{l + \pi f d_m \sec \alpha}{\pi d_m - f l \sec \alpha} \right) + \frac{F_i f_c d_c}{2} \quad (a)$$

- Applying $\tan l = l/\rho d_m$,

$$T = \frac{F_i d_m}{2} \left(\frac{\tan \lambda + f \sec \alpha}{1 - f \tan \lambda \sec \alpha} \right) + \frac{F_i f_c d_c}{2} \quad (b)$$

- Assuming a washer face diameter of $1.5d$, the collar diameter is $d_c = (d + 1.5d)/2 = 1.25d$, giving

$$T = \left[\left(\frac{d_m}{2d} \right) \left(\frac{\tan \lambda + f \sec \alpha}{1 - f \tan \lambda \sec \alpha} \right) + 0.625 f_c \right] F_i d \quad (c)$$

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Relating Bolt Torque to Bolt Tension

$$T = \left[\left(\frac{d_m}{2d} \right) \left(\frac{\tan \lambda + f \sec \alpha}{1 - f \tan \lambda \sec \alpha} \right) + 0.625 f_c \right] F_i d \quad (c)$$

- Define term in brackets as *torque coefficient* K

$$K = \left(\frac{d_m}{2d} \right) \left(\frac{\tan \lambda + f \sec \alpha}{1 - f \tan \lambda \sec \alpha} \right) + 0.625 f_c \quad (8-26)$$

$$T = K F_i d \quad (8-27)$$

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Typical Values for Torque Coefficient K

$$T = K F_i d \tag{8-27}$$

- Some recommended values for K for various bolt finishes is given in Table 8–15
- Use $K = 0.2$ for other cases

Table 8–15

Torque Factors K for Use with Eq. (8–27)

Bolt Condition	K
Nonplated, black finish	0.30
Zinc-plated	0.20
Lubricated	0.18
Cadmium-plated	0.16
With Bowman Anti-Seize	0.12
With Bowman-Grip nuts	0.09

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Recommended Preload

$$F_i = \begin{cases} 0.75F_p & \text{for nonpermanent connections, reused fasteners} \\ 0.90F_p & \text{for permanent connections} \end{cases} \tag{8-31}$$

$$F_p = A_t S_p \tag{8-32}$$

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Gasketed Joints

- For a full gasket compressed between members of a bolted joint, the gasket pressure p is found by dividing the force in the member by the gasket area per bolt.

$$p = -\frac{F_m}{A_g/N} \quad (a)$$

- The force in the member, including a load factor n ,

$$F_m = (1 - C)nP - F_i \quad (b)$$

- Thus the gasket pressure is

$$p = [F_i - nP(1 - C)] \frac{N}{A_g} \quad (8-33)$$

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Gasketed Joints

- Uniformity of pressure on the gasket is important
- Adjacent bolts should no more than six nominal diameters apart on the bolt circle
- For wrench clearance, bolts should be at least three diameters apart
- This gives a rough rule for bolt spacing around a bolt circle of diameter D_b

$$3 \leq \frac{\pi D_b}{Nd} \leq 6 \quad (8-34)$$

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Fatigue Stresses

- With an external load on a per bolt basis fluctuating between P_{min} and P_{max} ,

$$F_{bmin} = CP_{min} + F_i \tag{a}$$

$$F_{bmax} = CP_{max} + F_i \tag{b}$$

$$\sigma_a = \frac{(F_{bmax} - F_{bmin})/2}{A_t} = \frac{(CP_{max} + F_i) - (CP_{min} + F_i)}{2A_t}$$

$$\sigma_a = \frac{C(P_{max} - P_{min})}{2A_t} \tag{8-35}$$

$$\sigma_m = \frac{(F_{bmax} + F_{bmin})/2}{A_t} = \frac{(CP_{max} + F_i) + (CP_{min} + F_i)}{2A_t}$$

$$\sigma_m = \frac{C(P_{max} + P_{min})}{2A_t} + \frac{F_i}{A_t} \tag{8-36}$$

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Bolted and Riveted Joints Loaded in Shear

- Shear loaded joints are handled the same for rivets, bolts, and pins
- Several failure modes are possible
 - (a) Joint loaded in shear
 - (b) Bending of bolt or members
 - (c) Shear of bolt
 - (d) Tensile failure of members
 - (e) Bearing stress on bolt or members
 - (f) Shear tear-out
 - (g) Tensile tear-out

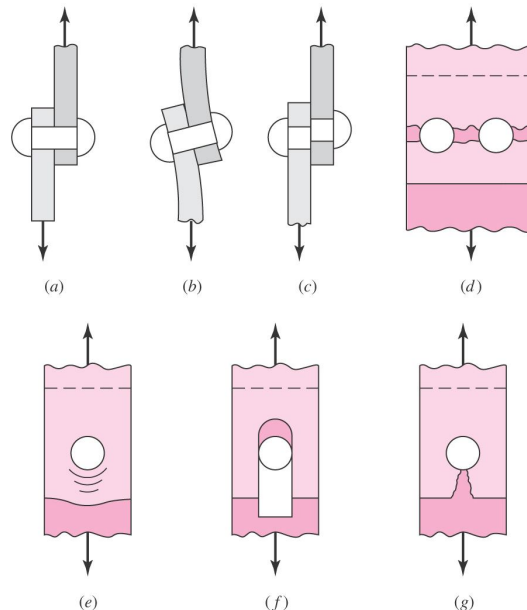


Fig. 8-23 *Shigley's Mechanical Engineering Design*

Failure by Bending

- Bending moment is approximately $M = Ft / 2$, where t is the grip length, i.e. the total thickness of the connected parts.
- Bending stress is determined by regular mechanics of materials approach, where I/c is for the weakest member or for the bolt(s).

$$\sigma = \frac{M}{I/c} \quad (8-52)$$



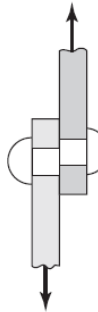
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Failure by Shear of Bolt

- Simple direct shear

$$\tau = \frac{F}{A} \quad (8-53)$$

- Use the total cross sectional area of bolts that are carrying the load.
- For bolts, determine whether the shear is across the nominal area or across threaded area. Use area based on nominal diameter or minor diameter, as appropriate.



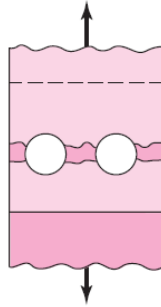
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Failure by Tensile Rupture of Member

- Simple tensile failure

$$\sigma = \frac{F}{A} \quad (8-54)$$

- Use the smallest net area of the member, with holes removed

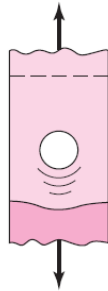


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Failure by Bearing Stress

- Failure by crushing known as *bearing stress*
- Bolt or member with lowest strength will crush first
- Load distribution on cylindrical surface is non-trivial
- Customary to assume uniform distribution over projected contact area, $A = td$
- t is the thickness of the thinnest plate and d is the bolt diameter

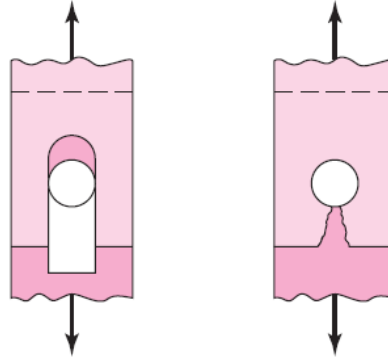
$$\sigma = -\frac{F}{A} \quad (8-55)$$



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Failure by Shear-out or Tear-out

- Edge shear-out or tear-out is avoided by spacing bolts at least 1.5 diameters away from the edge



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Shear Joints with Eccentric Loading

- *Eccentric* loading is when the load does not pass along a line of symmetry of the fasteners.
- Requires finding moment about centroid of bolt pattern
- Centroid location

$$\bar{x} = \frac{A_1x_1 + A_2x_2 + A_3x_3 + A_4x_4 + A_5x_5}{A_1 + A_2 + A_3 + A_4 + A_5} = \frac{\sum_1^n A_i x_i}{\sum_1^n A_i} \tag{8-56}$$

$$\bar{y} = \frac{A_1y_1 + A_2y_2 + A_3y_3 + A_4y_4 + A_5y_5}{A_1 + A_2 + A_3 + A_4 + A_5} = \frac{\sum_1^n A_i y_i}{\sum_1^n A_i}$$

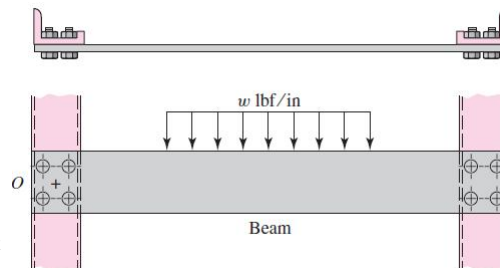


Fig. 8-27a

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